

# Graphical Computation of Jupiter's Perturbation on Asteroids<sup>1)</sup>

By

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## Abstract

For the computation of Jupiter's perturbation on Asteroids, a graphical method is recommended. It promises the computer easiness and sufficient accuracy for the derivation of data referred to the geometrical relation between Jupiter and Asteroids. The scale of the graph is a moderate one such that the orbit of Jupiter just fits on a sheet of typewriter paper.

Subsequent computation may be promptly carried out with a slide rule.

The coefficients of orbit improvement are also derived by a graphical method.

The incessant increase of the number of asteroids has compelled the "Astronomischen Rechen-Institut" at Berlin, Germany, to ask voluntary collaborations for the computation of Jupiter's perturbation on Asteroids. And Dr. Stracke of the Rechen-Institut has published instructions<sup>2)</sup> for the approximate calculation of special perturbation and for the improvement of orbital elements.

His method is that of the Variation of Elements which is generally treated in text books of theoretical astronomy, but Dr. Stracke has attained much simplification by using only three figure logarithms.

In view of the simplification, the computation also can be done with much easiness and moreover with sufficient accuracy by getting the necessary data for calculation with *graphical method* and by using a slide rule for the later computation. The scale of the graph may be a moderate one, such that the orbit of Jupiter can be graphed on a single sheet of typewriter paper.

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1. An abstract of a part of this paper was read at the Annual Meeting of the Mathematico-Physical Society of Japan, Kyoto, 1931 October.

2. G. Stracke: Veröffentlichungen des Astronomischen Rechen-Instituts zu Berlin, Nr. 44.

The weak point of the graphical method generally lies in its inaccuracy, but here the required accuracy is within its limit. And without this defect, it is much superior to numerical calculation in a few points, especially where the geometrical transformation of coordinates is concerned. Of all merits of the graphical method, the fact that it is carried out in parallel, so that accidental mistakes are to be detected easily, should receive special mention.

In the following treatment the same notations are used as those given in Dr. Stracke's paper.

Elements :

$M$	Mean Anomaly	$E$	Eccentric Anomaly
$M_0$	Mean Anomaly at the instant $t_0$	$v$	True Anomaly
$\omega$	Argument of Perihelion	$r$	Radius vector
$\Omega$	Longitude of Ascending Node	$p$	Parameter of Orbit
$i$	Inclination of Orbit to the Ecliptic	$u$	Argument of Latitude
$\varphi$	Angle of Eccentricity	$L$	Mean Longitude
$\mu$	Mean daily motion	$\pi$	Longitude of Perihelion
$a$	Semi-major axis	$k$	Gauss's constant in degree
	$\lambda_1$	$w$	Time interval
	$\beta_1$		Heliocentric longitude of Jupiter
	$r_1$		latitude " "
	$m_1$		Radius vector of Jupiter
	$S, T, W$		Mass of Jupiter in unit of Sun's mass
			the components of the perturbing force

Then we have,

$$1) \quad M = M_0 + \mu(t - t_0) = L - \pi$$

$$2) \quad E - \sin \varphi \sin E = M$$

$$3) \quad \begin{cases} r \sin v = a \cos \varphi \sin E \\ r \cos v = a(\cos E - \sin \varphi) \end{cases}$$

$$4) \quad a = \left( \frac{k}{\mu} \right)^{\frac{2}{3}}$$

$$5) \quad u = v + \omega = v + \pi - \Omega$$

$$6) \quad p = a \cos^3 \varphi$$

$$7) \quad \begin{cases} \cos B_1 \cos L_1 = \cos \beta_1 \cos(\lambda_1 - \Omega) \\ \cos B_1 \sin L_1 = \sin i \sin \beta_1 + \cos i \cos \beta_1 \sin(\lambda_1 - \Omega) \\ \sin B_1 = \cos i \sin \beta_1 - \sin i \cos \beta_1 \sin(\lambda_1 - \Omega) \end{cases}$$

$$\begin{aligned}
 8) \quad & \begin{cases} \xi_1 = r_1 \cos B_1 \cos(L_1 - u) \\ \eta_1 = r_1 \cos B_1 \sin(L_1 - u) \\ \zeta_1 = r_1 \sin B_1 \end{cases} \\
 9) \quad & \rho_1^2 = r_1^2 + r^2 - 2r\xi_1 \\
 10) \quad & K = \frac{wk m_1}{\sqrt{p}} \left( \frac{1}{\rho_1^3} - \frac{1}{r_1^3} \right) \\
 11) \quad & \begin{cases} S = K\xi_1 - \frac{wk m_1}{\sqrt{p}} \frac{r}{\rho_1^3} \\ T = K\eta_1 \\ W = K\zeta_1 \end{cases} \\
 12) \quad & \begin{cases} w \frac{di}{dt} = r \cos u. W \\ w \frac{d\Omega}{dt} = \frac{r \sin u}{\sin i}. W \\ w^2 \frac{d\mu}{dt} = -\frac{3k\tau w}{\sqrt{a}} \sin \varphi \sin v. S - \frac{3k\tau w}{\sqrt{a}} \frac{p}{r}. T \\ w \frac{d\pi}{dt} = -\frac{p}{\sin \varphi} \cos v. S + \frac{p+r}{\sin \varphi} \sin v. T + \tan \frac{i}{2} r \sin u. W \\ w \frac{dL}{dt} = -p \tan \frac{\varphi}{2} \cos v. S - 2 \cos \varphi r. S \\ \quad \quad \quad + \tan \frac{\varphi}{2} (p+r) \sin v. T + \tan \frac{i}{2} r \sin u. W \\ w \frac{d\varphi}{dt} = a \cos \varphi \sin v. S + a \cos \varphi \left[ \sin \varphi + \cos v \left( 1 + \frac{r}{a} \right) \right]. T \end{cases}
 \end{aligned}$$

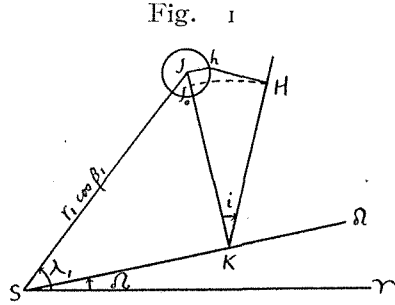
### Relative Positions of Jupiter and the Asteroid

We take the orbital plane of the asteroid as the fundamental plane of reference. Then the position of Jupiter can be represented by a point of orthogonal projection to the plane and the elevation above it.

For a while, considering the plane to be the ecliptic (in Fig. 1), let  $S$  be the position of the Sun, and a line  $SY$  be in the direction of the Vernal equinox. Draw a line  $SJ$ , making an angle  $\lambda_1$  with the initial line  $SY$ , and measure a curtate radius  $r_1 \cos \beta_1$  from  $S$  to  $J$ , then  $J$  is the projected position of Jupiter in the ecliptic, and it should be

understood that Jupiter lies just above that point as far as  $r_1 \sin \beta_1$ . If we describe a small circle  $h$  with a radius  $r_1 \sin \beta_1$  about the point  $J$ , the elevation above the ecliptic (or the depression according to the case) may be clearly represented. For the sake of convenience of operation,  $r_1 \cos \beta_1$  and  $r_1 \sin \beta_1$  (in natural numbers) should be tabulated beforehand.

Next, draw the nodal line  $S\Omega$  passing through the Sun, making an angle  $\Omega$  with  $SY$ , then we notice



that the part of Jupiter's orbit, or strictly speaking, the part of the ecliptic which lies on the side of this line into which the asteroid passes the Ascending node, is below the fundamental plane. If we draw a normal  $JK$  from the point  $J$  to  $S\Omega$ , intersecting at  $K$  with it, we have

$$13) \quad \begin{cases} JK = r_1 \cos \beta_1 \sin(\lambda_1 - \Omega) \\ SK = r_1 \cos \beta_1 \cos(\lambda_1 - \Omega). \end{cases}$$

Now let  $Jh$  be a radius of the small circle above mentioned, drawn parallel and in the same or opposite direction with  $S\Omega$ , according to whether Jupiter lies above or below the ecliptic, i. e.  $r_1 \sin \beta_1$  is positive or negative.

Make a base line  $KH$  through the foot  $K$  and at an angle  $i$  with the line  $JK$  in the clockwise direction, and then draw a line perpendicularly to the base line  $KH$  from the point  $h$ . If we denote the intersection point with  $H$ , then we will see from the figure 1 that the distance  $KH$  is equal to  $JK \cos i + Jh \sin i$ , or

$$14) \quad KH = r_1 \cos i \cos \beta_1 \sin(\lambda_1 - \Omega) + r_1 \sin i \sin \beta_1$$

and also that the perpendicular distance  $hH$  is equal to  $JK \sin i - Jh \cos i$ . And it will easily be seen that this distance is the depression of Jupiter under the fundamental plane, but since we measure the height of Jupiter above the fundamental plane positively, we must change the sign and we have

$$15) \quad hH = r_1 \cos i \sin \beta_1 - r_1 \sin i \cos \beta_1 \sin(\lambda_1 - \Omega).$$

In the graphical method, the elevation or depression can be easily visualized by measuring the height from  $H$  to  $h$  positively towards  $S$ .

Now, if we describe a small arc with a radius  $KH$  and its centre

at  $K$ , cutting the line  $KJ$  at a point  $J_0$ , then the point  $J_0$  represents the projected position of Jupiter on the fundamental plane. And join  $J_0$  to  $S$ , then the angle  $\Omega SJ_0$  is nothing but the longitude in the fundamental plane. If we compare the equations (13), (14) and (15) with the equations (7), we will find easily that the angle has to be designated by  $L_1$  and also that

$$SJ_0 = r_1 \cos B_1$$

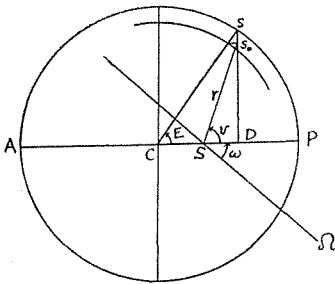
$$hH = r_1 \sin B_1.$$

The position of the asteroid also can be drawn graphically. For that purpose, first compute the mean anomaly  $M$  for an epoch  $t$  by the formula (1) and then require the eccentric anomaly  $E$ , where it is sufficient to get  $E$  as accurate as one-tenth of a degree.

This process may be carried out very conveniently by the use of the Keplerian which gives the mechanical solution of Kepler's equation<sup>1)</sup>.

Now draw the apside line  $ASP$  (Fig. 2) through the Sun, making an angle  $\omega$  with the nodal line  $SQ$ , and denote the Perihelion point and the Aphelion point by  $P$  and  $A$  which are distant from  $S$  by  $a(1 - \sin\varphi)$ ,  $a(1 + \sin\varphi)$  respectively.

Fig. 2



Then if we describe a circle with  $AP$  as its diameter, we can get the position of the asteroid in the following manner.

Take a point  $s$  on the periphery of the circle  $PsA$ , subtending an angle  $PCs$  at its centre  $C$  equal to  $E$ , and draw a normal to the apside line  $ASP$  passing the point  $s$ , and denote the foot of the normal by  $D$ . Now take a point of  $s_0$  in the line  $sD$  dividing the normal in the ratio such that  $sD : s_0D = 1 : \cos\varphi$ , then  $s_0$  is the true position of the asteroid in the orbit. Consequently, the distance  $Ss_0$  is the radius vector  $r$ , and the angle  $PSs_0$  is the true anomaly  $v$ .

$$\begin{cases} s_0D = r \sin v \\ SD = r \cos v \end{cases}$$

1. M. N. Vol. 87 p. 207.

and the angle measured from  $SQ$  should be denoted by  $u$ .

**Necessary data for the computation**

In the figure 3, if we get a line  $J_0Q$  passing through the point  $J_0$  perpendicular to the line  $Ss_0$ , then we have

$$16) \quad \begin{cases} SQ = SJ_0 \cos(L_1 - u) = r_1 \cos B_1 \cos(L_1 - u) \\ J_0Q = SJ_0 \sin(L_1 - u) = r_1 \cos B_1 \sin(L_1 - u). \end{cases}$$

And remembering the equations (8), we have the following relations :

$$\begin{cases} \xi_1 = SQ \\ \eta_1 = J_0Q \\ \zeta_1 = hH \end{cases}$$

where  $\xi_1$  must be measured positively towards  $s_0$ ; and  $\eta_1$  is positive when  $J_0$  proceeds  $s_0$  in longitude i. e.  $J_0$  lies in the left side of the direction  $Ss_0$ ; about the sense of  $\zeta_1$ , it has been already stated. To get the distance between Jupiter and the asteroid, first join the points  $J_0$  and  $s_0$ , then the distance  $J_0s_0$  expresses the projected distance in the fundamental plane. Now measure a distance  $J_0N$  equal to  $hH$  in (15) from  $J_0$  perpendicularly to the line  $J_0s_0$  and join the points  $N$  and  $s_0$ , then  $Ns_0$  is the required distance  $\rho_1$ . Because it enters into the computation by the form of inverse cube, it is convenient to have a table of the inverse cube with the argument of  $\rho_1$ .

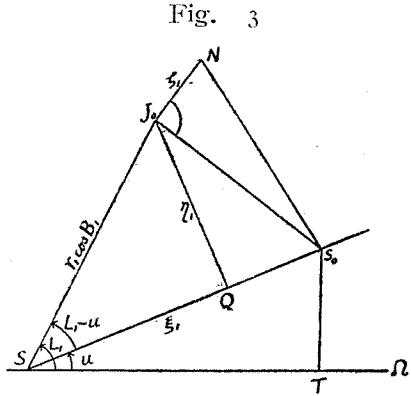


Fig. 3

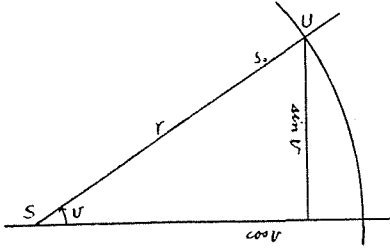
Other values necessary for the computation can be arrived at in the following manner :

Drawing a perpendicular  $s_0T$  from the point  $s_0$  to the nodal line, measure the length of the perpendicular  $s_0T$ , and the distance from the Sun to the foot of the normal,  $ST$ ; they give  $r \sin u$  and  $r \cos u$  respectively.

$s_0T$  is positive when  $s_0$  lies in the left side of the direction  $SQ$ , and  $ST$  must be measured positively along the direction of  $SQ$ .

Next, (Fig. 4) draw a circle around  $S$  with a unit radius, then it will cut the line  $Ss_0$  or its elongation at  $U$ .

Fig. 4



Then referring to the apside line, measure the normal and its offset in exactly the same way. They are nothing but  $\sin \nu$  and  $\cos \nu$  respectively.

**Formulae of Perturbation**

The formulae of perturbation (12) are rearranged for the convenience of the computation as follows; the modified form will be more profitable for the use of the slide rule, provided that the value of  $\varphi$  is not especially small.

$$K = \frac{wkm_1}{\sqrt{p}} \left( \frac{1}{\rho_1^3} - \frac{1}{r_1^3} \right)$$

$$wkm_1 = 75.28$$

$$\text{for } w = 80$$

$$\begin{cases} S = K\xi_1 - \frac{wkm_1}{\sqrt{p}} \frac{r}{\rho_1^3} \\ T = K\eta_1 \\ W = K\zeta_1 \end{cases}$$

$$\begin{cases} \delta i = r \cos u \cdot W \\ \delta \Omega = \frac{r \sin u}{\sin i} \cdot W \end{cases}$$

$$k_2 = 41.29$$

$$\begin{cases} 80\delta\mu_1 = -\frac{k_2 \sin\varphi}{\sqrt{a}} \sin\nu \cdot S \\ 80\delta\mu_2 = -\frac{k_2 p}{\sqrt{a}} \frac{1}{r} \cdot T \end{cases}$$

$$\begin{cases} \delta\varphi_0 = -\frac{1}{\tan\varphi} r \cdot T \\ \delta\varphi_1 = -\frac{a^{\frac{3}{2}}}{k_2 \tan\varphi} \cdot 80\delta\mu_1 \\ \delta\varphi_2 = -\frac{a^{\frac{3}{2}}}{k_2 \tan\varphi} \cdot 80\delta\mu_2 \end{cases}$$

$$\begin{cases} \delta\pi_1 = -\frac{p}{\sin\varphi} \cos\nu \cdot S \\ \delta\pi_2 = \frac{1}{\sin\varphi} (p+r) \sin\nu \cdot T \\ \delta\pi_3 = 2\sin^2 \frac{i}{2} \cdot \delta\Omega \end{cases}$$

$$\begin{cases} \delta L_0 = -2\cos\varphi \cdot r \cdot S \\ \delta L_1 = 2\sin^2 \frac{\varphi}{2} \cdot \delta\pi_1 \\ \delta L_2 = 2\sin^2 \frac{\varphi}{2} \cdot \delta\pi_2 \\ \delta L_3 = \delta\pi_3 \end{cases}$$

## Example

Here is given an example of computation. The scale of the graph was such that the astronomical unit was represented in 2 centimeters, so the orbit of Jupiter fits on to a sheet of typewriter paper. The angles were measured with a celluloid protractor of ordinary size of 7 centimeters radius, consequently  $E$  and  $\lambda_1$  were plotted within one tenth of a degree of accuracy.

For the purpose of comparison with Dr. Stracke's method, the same example as given in the "Veröffentlichungen" was taken :

(694) Ekard

$t_0 = 1909$  Dec. 3.5 U. T.

$M_0 = 46.350$

$$\left. \begin{array}{l} \omega = 108.243 \\ \Omega = 231.414 \\ i = 15.756 \end{array} \right\} 1910.0$$

$\varphi = 18.867$

$\mu = 813''.30 = 0''.22592$

$\log a = 0.42650$

and the perturbation of Jupiter was calculated for the same interval, from the initial epoch up to 1922 Dec. 15.5 U. T.

The computation scheme follows in the pages 56-60, where we find it convenient to arrange the data in columns instead of in rows. Similarly to the example given by Dr. Stracke, the starting elements were not used all through the whole interval, from 1910 Jan. 12.5 to 1922 Dec. 15.5, but the elements were improved twice, namely first on 1915 Jan. 26.5 and next on 1920 Feb. 9.5. The corrections of the elements are calculated in the following way :

$$\Delta M = \Delta L_\mu + \Delta L - \Delta \pi$$

$$\Delta \omega = \Delta \pi - \Delta \Omega$$

$$M = M_0 + \mu^0(t - t_0) + \Delta M$$

$$\Delta a = - \frac{a}{120} \frac{80 \Delta \mu^0}{\mu^0}$$

If we can use the summation table instead of the integration, then the value of  $\Delta L_\mu$ , corresponding to the first date of computation, will be



$$\Delta L_u(a) = {}''F(a) + \frac{1}{12}F'(a) - \frac{1}{240}F''(a) + \dots$$

and since

$${}''F(a) = \frac{1}{24}F(a - \tau v) + \dots$$

$$= \frac{1}{24}F(a) \text{ nearly,}$$

then

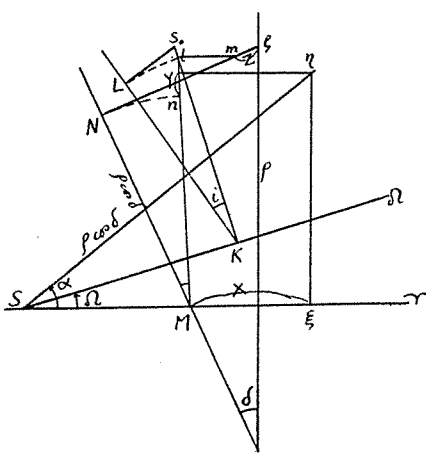
$$\Delta L_u(a) = \frac{1}{8}F(a).$$

### Comparison with the Observations

For the comparison with the observations, we can get the ephemerides with a graphical method. But in this case the graphical method is a little short, and a larger scale should be used to obtain more accuracy. It may not be out of place here to show the graphical method of getting the Right Ascension and Declination. Moreover, for the subsequent computation of the coefficients of the improvement of orbit, we can again take advantage of the graphical computation.

First, lay out a radius vector  $Ss_0$  from the Sun  $S$  (Fig. 5) subtending an angle  $u = v + \omega$  with the Nodal line  $S\Omega$ , and draw a normal  $s_0K$  to the line  $S\Omega$  from the position of the asteroid  $s_0$ .

Fig. 5



Then measure an angle  $i$  at  $K$  in the counter-clockwise direction and drop a normal from  $s_0$  to the base line, let the foot be denoted by  $L$ . If we take a point  $l$  on the line  $Ks_0$ , such as

$$Kl = KL$$

then the point  $l$  is the projected position of the asteroid in the plane of the ecliptic. From  $l$ , draw a line  $LM$  perpendicularly to the equinoctial line  $ST$ , and, in turn, another line  $lm$  perpendicularly to  $LM$ , and make  $lm = Ls_0$ .

$lm$  is to be drawn in the direction of  $ST$ , when  $s_0$  is in the side of  $\Omega$  referred to  $LK$ , and if otherwise to the opposite.

Next, pass a line  $MN$  through  $M$ , making an angle  $\epsilon$  in the counter-clockwise direction with  $MI$ , then if we draw a normal from the point  $m$  to the line intersecting at  $N$  with it, we have

$$\begin{cases} x = SM \\ y = MN \\ z = Nm. \end{cases}$$

$SM$  is measured as positive when  $M$  lies in the direction  $SY$ ,  $MN$  is positive when  $N$  lies in the lefthand side of the direction from  $S$  to  $Y$ .  $Nm$  is positive when  $m$  is in the same side with  $Y$  referred to  $MN$ .

When we add  $X$ ,  $Y$ , and  $Z$  to these values of  $x$ ,  $y$  and  $z$  respectively, we get  $\rho \cos \delta \cos \alpha$ ,  $\rho \cos \delta \sin \alpha$ ,  $\rho \sin \delta$ , and the subsequent process to get  $\alpha$ ,  $\delta$  and  $\rho$  is easily seen from the figure.

### Computation of the factors of "Bahnverbesserung".

When we are given with  $t-t_0$ ,  $M$ ,  $E$  and the orbital elements, we get graphically  $r$ ,  $\sin v$ , so we can compute the coefficients  $l$ ,  $m$ ,  $n$ , and  $o$  with a slide rule by the following formulae:

$$\begin{cases} l = \frac{a \cos \varphi \frac{a}{r} (t-t_0)}{3600} \cdot \frac{1}{\rho} = \frac{t-t_0}{3600} o \\ m = \frac{(\dot{\phi} + r) \sin v}{\cos \varphi} \cdot \frac{1}{\rho} \\ n = \frac{r}{\rho} \\ o = a \cos \varphi \frac{a}{r} \cdot \frac{1}{\rho} \end{cases}$$

$\sin F$  and  $\cos F$  which are necessary for the computation of the absolute term can be derived graphically by the following process:

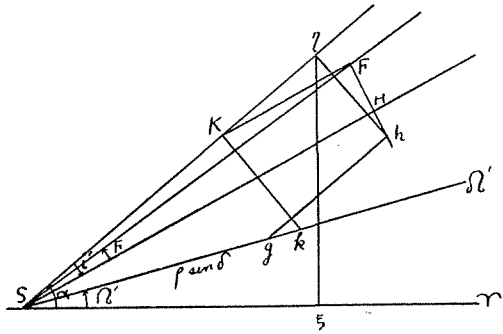
In Fig. 6, taking a point  $g$  on  $SQ'$ , distant from the Sun by  $\rho \sin \delta$ , close a rectangle with  $g\eta$  as its diagonal and with a part of  $S\eta$  as one side. Denote the angular point by  $h$ . Here we must be careful to measure the length  $Sg$  in the opposite direction when  $\rho \sin \delta$  is negative. Next, pass a base line through  $S$ , making an angle  $i'$  with  $S\eta$  in the clockwise direction, and project the point  $h$  to the base line, then the offset  $SH$  of the normal gives

$$\rho \cos i' \cos \delta + \rho \sin i' \sin \delta \sin(\alpha - \Omega')$$

which is expressed by  $\rho \cos g \cos F$ .

Again, taking a point  $k$  on  $SQ'$ , distant from the Sun by  $\rho$ , drop a normal  $kK$  to the line  $S\gamma$  through  $k$ . Subsequently, make a line parallel to  $SH$  through the point  $K$  and let it intersect with the line  $hH$  or its elongation. If we denote the intersection point with  $F$ , then the angle  $HSF$  is the required angle  $F$ .  $F$  is measured positively in the counter-clockwise direction from  $SH$ .

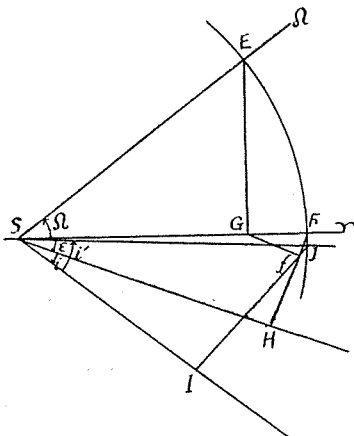
Fig. 6



The process may be carried out as the continuation of the preceding article. In that occasion, we must be careful to distinguish  $\Omega'$ ,  $i'$  from  $\Omega$ ,  $i$ , the former set of which are the Right Ascension of the Ascending Node and Inclination of the orbit referred to the equator and the latter are those referred to the ecliptic.

To correct the position of the nodal line, the best way is first to find  $i'$ , the inclination of the orbit to the equator.

Fig. 7



Describe an arc of circle of a unit radius, having its centre at  $S$ , and cut the nodal line  $SQ$  and the Vernal line  $SY$  at  $E$  and  $F$  respectively. Next draw a normal through  $E$  to the Vernal line  $SY$ , and let it be intersected with it at  $G$ .

Then subtend two lines  $SH$ ,  $SI$ , making angles  $\epsilon$ ,  $\epsilon + i'$  respectively with  $SY$  at  $S$ . Now drop a normal through  $F$  to the line  $SH$  intersecting at  $H$ , and drawing a line through  $G$  and parallel to  $SH$ , cut  $FH$  at  $f$ .

And subsequently, a normal to the line  $SI$  through  $f$  is to be drawn.

Then  $SI$  gives  $\cos i'$ , so if we cut the line  $If$  or its elongation at  $J$  with a unit radius from  $S$ , then the angle  $JSI$  is equal to  $i'$ .

Referring to the same diagram, but to Fig. 8, draw a line  $GK$

through the point  $G$  making an angle  $\epsilon$  with  $GE$ . Drop a normal from the point  $E$  to  $GK$ , intersecting at  $K$ . If we draw a line  $EN$  through  $E$ , making an angle  $90-i'$  with the normal  $EK$  to the side of  $G$ , then it will cut the base line  $GK$  or its elongation at  $N$ . Then  $NK$  gives the equatorial distance (say) between the nodes of the equator and of the ecliptic. The alternative to get the length  $NK$  is to obtain the base line corresponding to a side, equal to  $EK$ , which subtends the angle  $JSI$  in Fig. 7.

The next step is thus: cut the normal  $GE$  at  $L$  with a radius equal to  $GK$ , and its centre at  $G$ . Then describe a circle, having  $SL$  as its diameter, and cut off a chord  $LM$  equal to  $KN$ , in the side nearer to  $Y$ , then the line joining  $M$  and  $S$  gives the position of the nodal line in the equator. It should be noted here that in such a case where the normal from  $E$  to the equinoctial line does not meet with it in the branch of  $SY$ , i. e. it falls on the Autumnal line  $SY'$ , then substitute the diametrical line of the nodal line, and proceed in the same way as above, except that  $SI$  (in Fig. 7) is to be laid so as to make an angle  $\epsilon-i'$ , instead of  $\epsilon+i'$ .

Following remarks should be given here, to cover special cases.

a) Although it is not generally the case for asteroids to have a large value of  $i'$ , there are sometimes cases where  $i'$  is so large that the normal from the point  $f$  (Fig. 7) does not meet with the line  $SI$ , then substitute the diametrical line of  $SI$ .

b) And when the normal from the point  $f$  to the line  $SI$  or its substitute, is averted from  $SH$  line, then we should measure the chord  $LM$  in the semi-circle of the further side with respect to  $Y$ .

c) With regard to the sense of the required nodal line, it should be noted that when  $\Omega$  lies in the first two quadrants, then  $\Omega'$  is to be taken either in the first or in the second quadrant, and when  $\Omega$  lies in the last two quadrants, then  $\Omega'$  is in the third or in the fourth quadrant.

d) When  $NK$  is equal to  $LS$  in length, the chord  $LM$  coincides with the diameter of the circle, and the position of the nodal line seems to be indeterminate, but it means that the nodal line is to be laid perpendicularly to the line  $LS$ . And when it is nearly the case, a small error in the length  $NK$  will affect the direction of  $\Omega'$  very much, so it is recommended to use another method for an alternative. The process is thus: In Fig. 7 and 9, close a rectangle with  $F, f$  and  $G$ , and denote the new angular point by  $g$ , and let the line  $SH$  be

elongated to intersect with the unit circle at  $h$ , then draw two normals  $gP$ ,  $hQ$  to  $SI$  through the points  $g$ ,  $h$  respectively.

If we cut fragments  $S\rho$  in  $SY$ , and  $Sq$  in  $S\Omega$ , respectively equal to  $gP$  and  $hQ$ , then the point  $o$  which is given by the intersection of the normal and the parallel to  $SF$  through  $p$  and  $q$ , will give the direction  $\Omega'$ , by joining  $S$ ,  $o$ .  $Sq$  should be measured always towards  $\Omega$ , but  $S\rho$  is measured towards  $Y$  for the value of  $i$ , as far as the line  $SI$  or its diametrical line passes the point  $g$ , starting from the line  $SH$ , and for the larger value of  $i$ ,  $S\rho$  is measured to the opposite.

Fig. 8

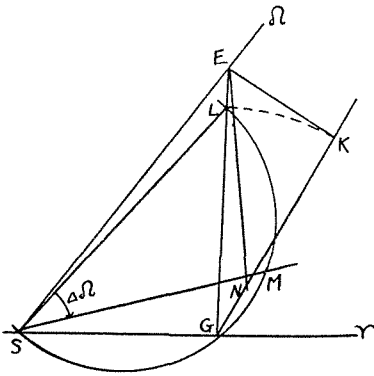
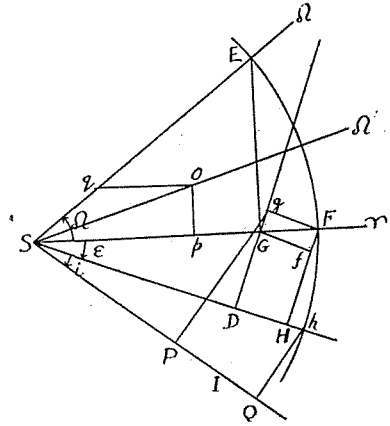


Fig. 9

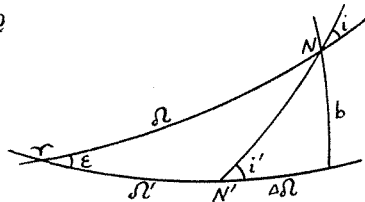


The above process will be approved by geometrical consideration, but may be verified in the following way.

Referring to the figure 10, we have the trigonometrical relations between these spacial elements as follows :

$$\begin{cases}
 \sin i' \sin \Omega' = \sin i \sin \Omega \\
 \sin i' \cos \Omega' = \cos i \sin \epsilon + \sin i \cos \epsilon \cos \Omega \\
 \cos i' = \cos i \cos \epsilon - \sin i \sin \epsilon \cos \Omega \\
 \sin \Delta \Omega = \tan b \cot i' \\
 \sin b = \sin \Omega \sin \epsilon
 \end{cases}
 \quad 17)$$

Fig. 10



When we refer to Figure 7

$$Y.S\Omega = \Omega$$

and as  $SE$ ,  $SF$  were taken as unit length,

$$\begin{aligned}
 SG &= \cos \Omega \\
 fH &= SG \sin \epsilon = \cos \Omega \sin \epsilon \\
 SH &= SF \cos \epsilon = \cos \epsilon \\
 SI &= SH \cos i - fH \sin i \\
 &= \cos i \cos \epsilon - \sin i \sin \epsilon \cos \Omega,
 \end{aligned}$$

so,  $SI = \cos i'$ , and since  $SJ = 1$

$$\angle JSI = i'.$$

And now we have, in Fig. 8

$$\begin{aligned}
 EG &= \sin \Omega \\
 EK &= EG \sin \epsilon = \sin \Omega \sin \epsilon \\
 KN &= EK \cot i' = \frac{\sin \Omega \sin \epsilon}{\tan i'} = LM \\
 GK &= GL = EG \cos \epsilon = \sin \Omega \cos \epsilon \\
 SG &= \cos \Omega
 \end{aligned}$$

Consequently,  $SL^2 = SG^2 + GL^2 = \cos^2 \Omega + \sin^2 \Omega \cos^2 \epsilon$   
and

$$1 - SL^2 = 1 - \cos^2 \Omega - \sin^2 \Omega \cos^2 \epsilon = \sin^2 \Omega \sin^2 \epsilon,$$

that is equal to  $\sin^2 b$  from the trigonometrical relation (17).

Thence, we have  $SL = \cos b$ .

In the rectangular triangle  $LSM$ ,

$$\sin LSM = \frac{LM}{SL} = \frac{\sin \Omega \sin \epsilon}{\tan i'} \frac{1}{\cos b} = \frac{\tan b}{\tan i'}$$

or equal to  $\sin \Delta \Omega$  from the same trigonometrical relation (17).

Consequently, we have

$$\angle LSM = \Delta \Omega$$

and  $SL$  is the projection of the nodal line  $SQ$  in the ecliptic to the plane of the equator, so we can see  $YSM$  corresponds to  $\Omega'$ .

In Fig. 9,

$$\begin{aligned}
 gD &= FH = SF \sin \epsilon = \sin \epsilon \\
 SD &= SG \cos \epsilon = \cos \Omega \cos \epsilon \\
 gP &= gD \cos i + SD \sin i = \cos i \sin \epsilon + \sin i \cos \epsilon \cos \Omega,
 \end{aligned}$$

so,  $gP = \sin i' \cos \Omega'$  from the equation (17).

$$\begin{aligned}
 Sq &= hQ = Sh \sin i = \sin i \\
 op &= Sq \sin \Omega = \sin i \sin \Omega = \sin i' \sin \Omega'
 \end{aligned}$$

consequently, we have

$$\angle oSp = \Omega'.$$

Example for the Computation of the Coefficients

Thorough comparison of the computation for all the dates with that given by Dr. Stracke seems to be unnecessary. Here we show the comparison of both the methods for a single date,

1913 Aug. 2.0

for which,

$t-t_0$	1337 <sup>d</sup> .5	
$M$	348 <sup>o</sup> .78	
$\omega$	108.00	}
$\Omega$	231.44	
$i$	15.75	
$\varphi$	18.83	
$a$	2.67	
$\varepsilon$	23 <sup>o</sup> .45	1913.0

are given.

We compute  $b=2.53$   $p=2.39$   $ae=0.86$   $E=-16^{\circ}.4$

and by the graphical construction, we read out the necessary data for computation as follows :

Dr. Stracke's computation

$r$	1.84		1.845
$\sin v$	-.389		-.387
$x$	+1.284		+1.2857
$y$	-1.316		-1.3213
$z$	-.032		-.0294
$\alpha$	317 <sup>o</sup> .3		317 <sup>o</sup> .18
$\delta$	+18.0		+17.88
$\rho$	.920		.925

and the subsequent computation with a slide rule gives,

$l$	+1.48		+1.47
$m$	-1.89		-1.87
$n$	+2.00		+2.00
$o$	+3.99		+3.96

With regard to  $F$ , Dr. Stracke accidentally mistook  $\Omega$ ,  $i$  as the elements referred to the ecliptic instead of as those referred to the equator.

But for the purpose of comparison, the same values of  $\Omega$ ,  $i$  were used, resulting the following values

Dr. Stracke's

$\sin F$	+.020		+.021
$\cos F$	+1.00		+1.000

## Tables

Data for the construction of the graph

Orbital constants		0.5 U. T.		M	E
	1	1910	1 12	55.39	73.1
	2		4 2	73.46	92.0
	3		6 21	91.53	109.0
	4		9 9	109.61	124.8
$a$ 2.670	5		11 28	127.68	139.7
$\sin\varphi$ .3234	6	1911	2 16	145.75	153.9
	7		5 7	163.82	167.8
$\cos\varphi$ .9463	8		7 26	181.90	-178.6
	9		10 14	199.97	-164.9
$ae$ .863	10	1912	1 2	218.04	-151.0
$b$ 2.53	11		3 22	236.12	-136.6
	12		6 10	254.19	-121.6
$p$ 2.39	13		8 29	272.26	-105.6
	14		11 17	290.34	-88.2
	15	1913	2 5	308.41	-68.9
	16		4 26	326.48	-47.1
	17		7 15	344.56	-22.6
	18		10 3	2.63	3.9
	19		12 22	20.70	30.0
	20	1914	3 12	38.77	53.7
	21		5 31	56.85	74.7
	22		8 19	74.92	93.4
	23		11 17	92.99	110.4
	24	1915	1 26	111.07	126.0

Factors for the subsequent computation

$\frac{k_1}{\sqrt{p}}$	48.70	$\frac{k_2 \sin\varphi}{\sqrt{a}}$	$\frac{1}{.1224}$
$\tan\varphi$	.3417	$\frac{k_2 p}{\sqrt{a}}$	60.40
$2 \cos\varphi$	$\frac{1}{.5284}$	$\frac{a^{\frac{3}{2}}}{k_2 \tan\varphi}$	.3092
$2 \sin^2 \frac{\varphi}{2}$	.0537	$\sin i$	.2715
$\frac{p}{\sin\varphi}$	$\frac{1}{.1353}$	$2 \sin^2 \frac{i}{2}$	.0375



	0.5 U. T.		$\rho_1$	$\xi_1$	$\eta_1$	$\zeta_1$	$r$	$r \sin u$	$r \cos u$	$\sin v$	$\cos v$
I	1910	1 12	6.76	-2.11	4.87	1.20	2.42	-.84	-2.28	1.000	-.039
2		4 2	6.52	-1.01	5.25	1.10	2.70	-1.69	-2.10	.937	-.351
3		6 21	6.30	-.18	5.38	.97	2.95	-2.39	-1.73	.811	-.583
4		9 9	6.05	.50	5.36	.83	3.16	-2.91	-1.22	.659	-.752
5		11 28	5.84	.99	5.30	.69	3.34	-3.26	-.65	.499	-.870
6	1911	2 16	5.60	1.45	5.21	.54	3.45	-3.45	-.05	.330	-.947
7		5 7	5.37	1.84	5.10	.40	3.51	-3.46	+.58	+.158	-.989
8		7 26	5.13	2.20	4.95	.22	3.54	-3.33	1.18	-.018	-1.000
9		10 14	4.87	2.53	4.77	.06	3.51	-3.05	1.73	-.190	-.983
10	1912	1 22	4.59	2.86	4.56	-.10	3.44	-2.65	2.18	-.360	-.936
11		3 2	4.29	3.20	4.28	-.27	3.31	-2.12	2.54	-.529	-.852
12		6 10	3.96	3.60	3.91	-.43	3.14	-1.48	2.77	-.692	-.727
13		8 29	3.61	4.03	3.40	-.60	2.92	-.73	2.82	-.841	-.547
14		11 17	3.32	4.50	2.68	-.74	2.66	+.05	2.67	-.955	-.300
15	1913	2 5	3.12	4.95	1.54	-.88	2.36	.83	2.23	-1.000	+.044
16		4 26	3.21	5.13	-.10	-1.02	2.08	1.50	1.47	-.890	.460
17		7 15	3.65	4.59	-2.18	-1.14	1.87	1.83	+.44	-.526	.855
18		10 3	4.45	2.79	-4.16	-1.24	1.80	1.65	-.73	+.102	.990
19		12 22	5.31	.47	-4.94	-1.32	1.93	.98	-1.65	.660	.752
20	1914	3 12	6.07	-1.40	-4.72	-1.38	2.16	+.05	-2.16	.943	+.339
21		5 31	6.66	-2.63	-4.10	-1.40	2.45	-.92	-2.28	1.000	-.066
22		8 19	7.12	-3.31	-3.50	-1.42	2.73	-1.76	-2.08	.928	-.375
23		11 17	7.50	-3.75	-3.00	-1.42	2.97	-2.45	-1.68	.797	-.603
24	1915	1 26	7.79	-4.03	-2.58	-1.41	3.18	-2.95	-1.16	+.643	-.764

	$\frac{I}{\rho_1^3}$	$\frac{I}{r_1^3}$	$\frac{I}{\rho_1^3} - \frac{I}{r_1^3}$	$K$	$K\xi_1$	$T = K\eta_1$	$W = K\zeta_1$	$\frac{I}{\sqrt{p}} \frac{r}{\rho_1^3}$	$S$
I	.00324	.00617	-.00293	-.1427	+.301	-.695	-.1712	.382	-.081
2	361	617	.256	-.1246	.1258	-.654	-.1370	.476	-.350
3	400	617	.217	-.1056	+.0190	-.568	-.1025	.575	-.556
4	452	618	.166	-.0809	-.0405	-.434	-.0672	.696	-.7365
5	502	620	.118	-.0575	-.0569	-.305	-.0397	.818	-.875
6	.00569	623	-.00054	-.0263	-.0372	-.137	-.0142	.9575	-.995
7	646	627	+.00019	+.0093	+.0171	+.0474	+.00372	1.106	-1.089
8	741	632	.109	.0531	.1167	.2624	.01166	1.276	-1.159
9	866	638	.228	.1110	.2809	.530	+.00667	1.479	-1.198
10	.01034	645	.389	.1894	.542	.864	-.01894	1.735	-1.193
11	1266	653	.613	.2984	.956	1.278	-.0807	2.044	-1.088
12	1610	662	.948	.462	1.66	1.804	-.1984	2.466	-.806
13	2126	672	1.454	.708	2.85	2.406	-.425	3.027	-.174
14	2733	682	2.051	.998	4.49	2.673	-.738	3.543	+.947
15	.03293	693	2600	1.266	6.27	+.1950	-1.113	3.787	2.483
16	3023	705	2318	1.128	5.80	-.1128	-1.151	3.067	2.732
17	2056	716	1340	.653	3.00	-1.423	-.745	1.870	+.1130
18	1135	729	+.00406	+.1975	+.551	-.8215	-.245	.995	-.444
19	668	741	-.00073	-.0355	-.0170	+.1753	+.0468	.628	-.645
20	.00447	753	306	-.149	+.2088	.7038	.2056	.471	-.262
21	338	765	427	-.208	.5473	.8532	.2912	.404	+.143
22	277	776	499	-.2432	.805	.8518	.3454	.369	.436
23	237	787	550	-.268	1.005	.805	.3808	.343	.662
24	.00212	.00796	-.00584	-.2844	+.1147	+.735	+.4015	.328	+.819

	$\delta z$	$\delta \Omega$	$\frac{I}{.1224}$ $80\delta\mu_1$	$-60.40$ $80\delta\mu_2$	$80\delta\mu$	$-3417$ $\delta\varphi_0$	$-3092$ $\delta(\varphi_1 + \varphi_2)$	$\delta\varphi$
1	+ .39	+ .53	+ .65	+ 17.35	+ 18.00	+ 4.92	- 5.56	- .64
2	.29	.85	2.68	14.63	17.31	5.17	- 5.35	- .18
3	.18	.90	3.68	11.63	15.31	4.91	- 4.74	+ .17
4	.08	.72	3.96	8.30	12.26	4.02	- 3.79	.22
5	.03	.48	3.57	5.52	9.09	2.98	- 2.81	+ .17
6	.00	+ .18	2.68	+ 2.40	5.08	+ 1.38	- 1.57	- .19
7	.00	-.05	+ 1.41	-.82	+ .59	-.49	-.18	-.67
8	.01	-.14	-.17	- 4.48	- 4.65	- 2.72	+ 1.44	- 1.28
9	+ .01	-.08	- 1.86	- 9.12	- 10.98	- 5.44	3.40	- 2.04
10	-.04	+ .19	- 3.51	- 15.16	- 18.67	- 8.70	5.77	- 2.93
11	-.25	.63	- 4.70	- 23.33	- 28.03	- 12.38	8.67	- 3.71
12	-.55	1.08	- 4.56	- 34.7	- 39.26	- 16.58	12.13	- 4.45
13	- 1.34	+ 1.14	- 1.19	- 49.8	- 50.99	- 20.60	15.74	- 4.86
14	- 1.97	-.14	+ 7.49	- 60.7	- 53.21	- 20.80	16.45	- 4.35
15	- 2.48	- 3.40	20.28	- 49.9	- 29.62	- 13.47	+ 9.16	- 4.31
16	- 1.69	- 6.37	19.84	+ 3.28	+ 23.12	+ .69	- 7.15	- 6.46
17	-.33	- 5.02	4.86	46.0	50.86	7.80	- 15.71	- 7.91
18	-.18	- 1.49	.37	+ 27.56	+ 27.93	+ 4.33	- 8.64	- 4.31
19	-.08	+ .17	3.47	- 5.48	- 2.01	-.99	+ .62	-.37
20	-.44	+ .04	+ 2.02	- 19.80	- 17.78	- 4.45	5.50	+ 1.05
21	-.66	-.09	- 1.17	- 21.02	- 22.19	- 6.12	6.86	.74
22	-.72	- 2.24	- 3.30	- 18.84	- 22.14	- 6.80	6.85	+ .05
23	-.64	- 3.44	- 4.32	- 16.36	- 20.68	- 7.00	6.40	- .60
24	-.46	- 4.36	- 4.31	- 13.95	- 18.26	- 6.84	+ 5.65	- 1.20

	$\frac{I}{.1353}$ $\delta\pi_1$	$p+r$	$(p+r)\sin v$	$.3234$ $\delta\pi_2$	$\delta(\pi_1 + \pi_2)$	$.0375$ $\delta\pi_3 = \delta L_0$	$\delta\pi$	$\frac{I}{.5284}$ $\delta L_0$	$.0537$ $\delta(L_1 + L_2)$	$\delta L$
1	-.02	4.81	+ 4.81	- 10.35	- 10.37	+ .02	- 10.35	+ .37	-.56	-.17
2	-.91	5.09	4.77	- 9.65	- 10.56	3	- 10.53	1.79	-.57	+ 1.25
3	- 2.40	5.34	4.33	- 7.01	- 10.01	3	- 9.98	3.11	-.54	2.60
4	- 4.09	5.55	3.66	- 4.91	- 9.00	3	- 8.97	4.41	-.48	3.96
5	- 5.62	5.73	2.86	- 2.70	- 8.32	2	- 8.31	5.53	-.45	5.10
6	- 6.96	5.84	1.93	-.82	- 7.78	+ .01	- 7.76	6.50	-.42	6.09
7	- 7.96	5.90	+ .933	+ .14	- 7.82	0	- 7.82	7.24	-.42	6.82
8	- 8.56	5.93	- .107	-.09	- 8.65	- .01	- 8.66	7.76	-.46	7.29
9	- 8.71	5.90	- 1.12	- 1.84	- 10.55	0	- 10.55	7.96	-.57	7.39
10	- 8.26	5.83	- 2.10	- 5.60	- 13.86	+ .01	- 13.85	7.78	-.74	7.05
11	- 6.85	5.70	- 3.02	- 11.92	- 18.77	2	- 18.75	6.82	- 1.01	5.83
12	- 4.33	5.53	- 3.83	- 21.38	- 25.71	4	- 25.67	4.80	- 1.38	+ 3.46
13	- .70	5.31	- 4.46	- 33.20	- 33.90	+ .04	- 33.86	+ .96	- 1.82	-.82
14	+ 2.10	5.05	- 4.82	- 39.80	- 37.70	0	- 37.70	- 4.77	- 2.02	- 6.79
15	-.81	4.75	- 4.75	- 28.60	- 29.41	- .13	- 29.54	- 11.10	- 1.58	- 12.81
16	- 9.28	4.47	- 3.98	+ 1.39	- 7.89	-.24	- 8.13	- 10.76	-.42	- 11.42
17	- 7.15	4.26	- 2.24	+ 9.86	+ 2.71	-.19	+ 2.52	- 4.01	+ .15	- 4.05
18	+ 3.27	4.19	+ .43	- 1.09	2.18	-.06	2.12	+ 1.51	.12	+ 1.57
19	3.58	4.32	2.85	+ 1.35	5.13	+ .01	5.14	2.36	.28	2.65
20	.66	4.55	4.29	9.35	10.01	0	10.01	+ 1.07	.54	+ 1.61
21	.07	4.84	4.84	12.76	12.83	-.04	12.79	-.66	.69	-.01
22	- 1.21	5.12	4.75	12.50	13.71	-.08	13.63	- 2.25	.74	- 1.59
23	- 2.95	5.36	4.27	10.63	13.60	- .13	13.47	- 3.73	.73	- 3.13
24	+ 4.62	5.57	+ 3.58	8.14	+ 12.76	-.16	+ 12.60	- 4.93	+ .69	- 4.40

Graphical Computation of Jupiter's Perturbation on Asteroids 59

12 <sup>h</sup> U.T.	80δ <sub>μ</sub>	80Δ <sub>μ</sub>	ΔL <sub>μ</sub>	δL	ΔL	δπ	Δπ	δρ	Δρ	δΩ	ΔΩ	δi'	Δi'	12 <sup>h</sup> U.T.
		0			°.000		°.000		°.000		°.000		°.000	12 3
														1909
														1910
1 12	18	18	*.0009	0	.000	-10	-.010	-1	-.001	1	.001	0	.000	2 21
4 2	17	35	.0027	1	.001	-11	-.021	0	-.001	1	.002	0	.000	5 12
6 21	15	50	.0062	3	.004	-10	-.031	0	-.001	1	.003	0	.000	7 31
9 9	12	62	.0112	4	.008	-9	-.040	0	-.001	1	.004	0	.000	10 19
11 28	9		.0174	5		-8		0		0		0		
														1911
2 16	5	71		6	.013	-8	-.048	0	-.001	0	.004	0	.000	1 7
5 7	1	76	.0245	7	.019	-8	-.056	0	-.001	0	.004	0	.000	3 28
7 26	-5	77	.0321	7	.026	-9	-.064	-1	-.002	0	.004	0	.000	6 16
10 14	-11	72	.0398	7	.033	-9	-.073	-1	-.003	0	.004	0	.000	9 4
		61	.0470	7	.040	-11	-.084	-2	-.005	0	.004	0	.000	11 23
														1912
1 2	-19			7		-14	-.098	-3		0		0		2 11
3 22	-28	42	.0531	6	.047	-19	-.098	-4	-.008	0	.004	0	.000	5 1
6 10	-39	14	.0573	3	.053	-26	-.117	-4	-.012	1	.005	-1	.000	7 20
8 29	-51	-25	.0587	-1	.056	-34	-.143	-5	-.016	1	.006	-1	-.001	10 8
11 17	-53	-76	.0562	-1	.055	-38	-.177	-4	-.021	0	.007	-2	-.002	12 27
		-129	.0486	-7	.048		-.215		-.025	0	.007		-.004	
														1913
2 5	-30			-13		-30	-.245	-4	-.029	-3	.004	-2	-.006	3 17
4 26	23	-159	.0357	-11	.035	-8	-.253	-6	-.035	-6	.002	-2	-.008	6 5
7 15	51	-136	.0198	-4	.024	3	-.250	-5	-.043	-5	.007	0	-.008	8 24
10 3	28	-85	.0062	2	.020	2	-.248	-4	-.047	-1	.007	0	-.008	11 12
12 22	-2	57	-.0023	3	.022	5		0		0	-.008	0		
			-.0080											1914
3 12	-18	59		2	.025	10	-.243	1	-.047	0	-.008	0	-.008	1 31
5 31	-22	77	-.0139	0	.027	13	-.233	1	-.046	0	-.008	-1	-.008	4 21
8 19	-22	99	-.0216	-2	.027	14	-.220	0	-.045	-1	-.009	-1	-.009	7 10
11 7	-21	-121	-.0315	-3	.025	13	-.206	-1	-.045	-2	-.011	-1	-.010	9 28
		-142	-.0436	-3	.022	13	-.193	-1	-.046	-3	-.014		-.011	12 17
														1915
1 26	-19	-161	-.0578	-4	.018	13	-.180	-1	-.047	-4	-.018	0	-.011	3 7
4 16	-16	-177	-.0739	-5	.013	12	-.168	-2	-.049	-5	-.023	0	-.011	5 26
7 5	-14	-191	-.0916	-6	.007	10	-.158	-2	-.051	-5	-.028	0	-.011	8 14
9 23	-12	-203	-.1107	-7	.000	9	-.149	-3	-.054	-5	-.033	0	-.011	11 2
12 12	-9		-.1310	-7		8		-3		-4		0		
														1916
3 1	-7	-212	-.1522	-7	-.007	6	-.141	-3	-.057	-4	-.037	1	-.011	1 21
5 20	-4	-219	-.1741	-7	-.014	5	-.135	-3	-.060	-3	-.041	1	-.010	4 10
8 8	-1	-223	-.1964	-7	-.021	4	-.130	-3	-.063	-2	-.044	1	-.009	6 29
10 27	3	-224	-.2188	-7	-.028	4	-.126	-3	-.066	-1	-.046	0	-.008	9 17
		-221		-7	-.035	4	-.122	-3	-.069	-1	-.047	0	-.008	12 6

\* Refer to the page 49.

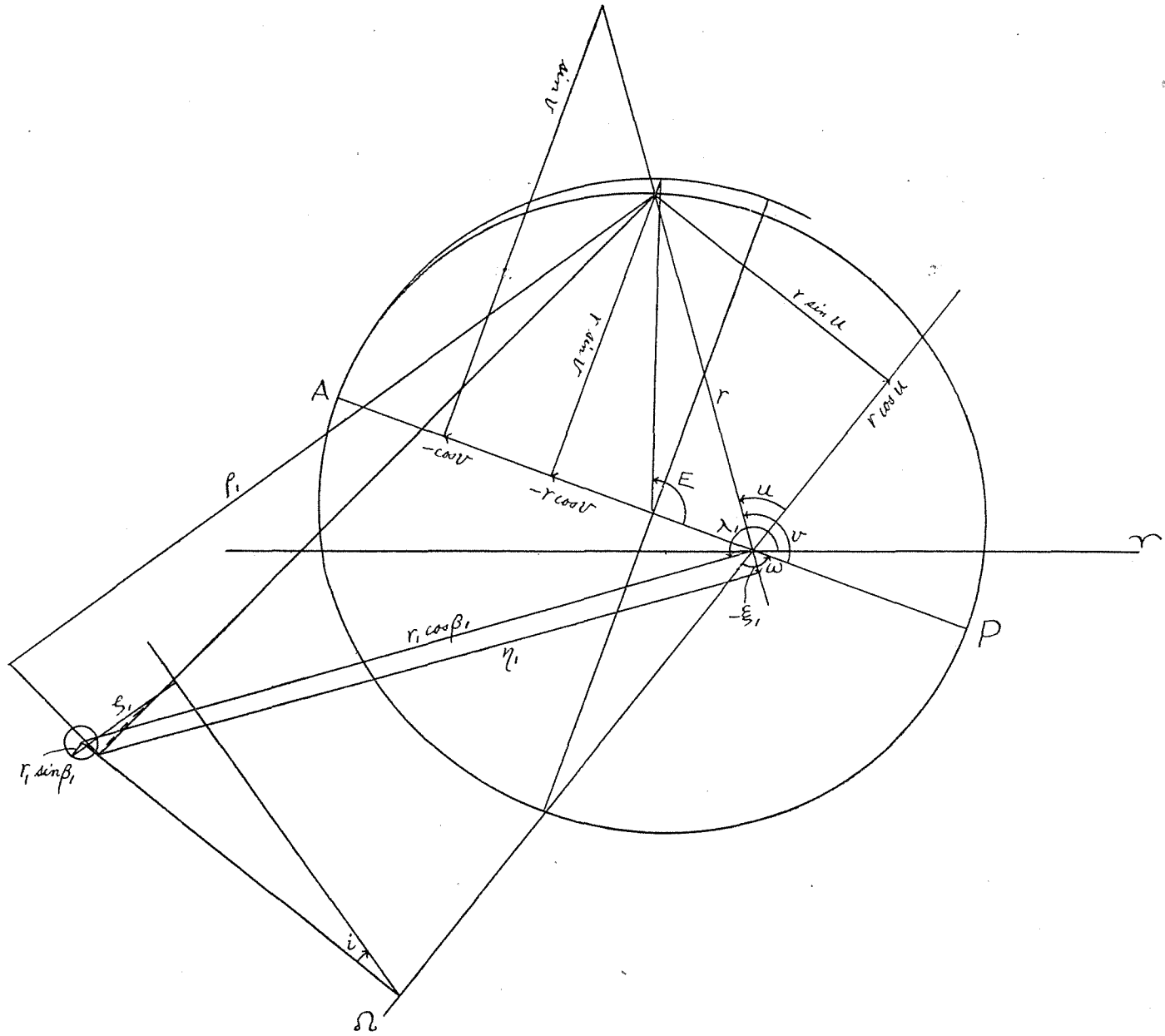
12 <sup>h</sup> U.T.	80δμ	80Δμ	ΔLμ	δL	ΔL	δπ	Δπ	δρ	Δρ	δΩ	ΔΩ	δz	Δz	12 <sup>h</sup> U.T.
1917														
1 15	8		°	-6	°	4	°	-3	°	0	°	0	°	2 24
4 5	13	-213	-.2409	-5	-.041	5	-.118	-3	-.072	0	-.047	0	-.008	5 15
6 24	19	-200	-.2622	-4	-.046	6	-.113	-3	-.075	0	-.047	0	-.008	8 13
9 12	25	-181	-.2822	-2	-.050	7	-.107	-3	-.078	0	-.047	0	-.008	10 22
12 1	24	-156	-.3003	0	-.052	5	-.100	-3	-.081	0	-.047	0	-.008	
			-.3159											
1918														
2 19	7	-132	-.3291	2	-.052	4	-.095	-1	-.084	0	-.047	0	-.008	1 10
5 10	-26	-125	-.3416	3	-.050	12	-.091	3	-.085	0	-.047	0	-.008	3 31
7 29	-59	-151	-.3567	-1	-.047	25	-.079	6	-.082	0	-.047	-1	-.008	6 19
10 17	-79	-210	-.3777	-10	-.048	37	-.054	7	-.076	-5	-.052	-3	-.009	9 7
		-289	-.3777		-.058		-.017		-.069		-.052		-.012	11 26
1919														
1 5	-68		-.4066	-22	-.080	35	.018	9	-.060	-17	-.069	-5	-.017	2 14
3 26	-29	-357	-.4423	-32	-.112	20	.038	12	-.048	-35	-.104	-6	-.023	5 5
6 14	15	-386	-.4809	-34	-.146	5	.043	16	-.032	-49	-.153	-5	-.028	7 24
9 2	46	-371	-.5180	-30	-.176	-2	.041	20	-.012	-56	-.209	-3	-.031	10 12
11 21	64	-325	-.5505	-23	-.199	1	.040	22	.010	-55	-.264	0	-.031	12 31
		-201	-.5505		-.199						-.264			
1920														
2 9	70	-191	-.5766	-15	-.214	4	.044	21	.031	-47	-.311	3	-.028	3 20
4 29	69	-122	-.5957	-10	-.224	12	.056	20	.051	-39	-.350	4	-.024	6 8
7 18	64	-58	-.6079	-5	-.229	19	.075	17	.068	-29	-.379	5	-.019	8 27
10 6	56	-2	-.6137	-1	-.230	23	.098	14	.082	-18	-.397	4	-.015	11 15
12 25	49		-.6139	1	-.230	26	.098	11	.082	-11	-.397	4	-.015	
1921														
3 15	39	47	-.6092	3	-.229	23	.124	7	.093	-5	-.408	-0.111		2 3
6 3	26	85	-.6007	4	-.226	18	.147	7	.100	-1	-.413	3	-.008	4 24
8 22	15	112	-.5895	4	-.222	12	.165	4	.104	0	-.414	2	-.006	7 13
11 10	1	127	-.5768	3	-.215	4	.177	2	.106	0	-.414	1	-.005	10 1
		128	-.5768		-.215		.181		.108		-.414		-.005	12 20
1922														
1 29	-12	116	-.5640	2	-.213	-1	.180	2	.110	0	-.414	0	-.005	3 10
4 19	-19	97	-.5524	0	-.213	-3	.177	3	.113	-1	-.415	0	-.005	5 29
7 8	-16	81	-.5427	-2	-.215	-2	.175	3	.116	0	-.416	0	-.005	8 17
9 26	-6	75	-.5346	-3	-.218	-3	.172	2	.118	0	-.416	0	-.005	11 5
12 15	3		-.5271	-3	-.218	-6	.172	1	.118	0	-.416	0	-.005	

Jupiter's Co-ordinates.

Date 12 <sup>h</sup> U.T.	$\lambda_1$	$r_1[\cos\beta_1]$	$r_1\sin\beta_1$	$\frac{10^5}{r_1^3}$	Date 12 <sup>h</sup> U.T.	$\lambda_1$	$r_1[\cos\beta_1]$	$r_1\sin\beta_1$	$\frac{10^5}{r_1^3}$
1910 1 12	183.8	5.45	+.12	617	1920 2 9	133.6	5.32	+.07	664
4 2	189.8	5.45	.12	617	4 29	140.0	5.34	.08	655
6 21	195.9	5.45	.12	617	7 18	146.2	5.37	.09	644
9 9	201.9	5.45	.12	618	10 6	152.4	5.39	.10	640
11 28	208.0	5.44	.12	620	12 25	158.6	5.40	.11	634
1911 2 16	214.1	5.43	+.11	623	1921 3 15	164.8	5.42	+.11	628
5 7	220.2	5.42	.11	627	6 3	170.9	5.43	.12	624
7 26	226.3	5.41	.10	632	8 22	177.0	5.44	.12	620
10 14	232.5	5.39	.09	638	11 10	183.0	5.45	.12	618
1912 1 2	238.7	5.37	+.08	645	1922 1 29	189.1	5.45	+.12	617
3 22	244.9	5.35	.07	653	4 19	195.1	5.45	.12	616
6 10	251.2	5.33	.06	662	7 8	201.2	5.45	.12	617
8 29	257.6	5.30	.04	672	9 26	207.2	5.45	.12	619
11 17	264.0	5.27	.03	682	12 15	213.3	5.44	.11	622
1913 2 5	270.5	5.24	+.02	693	1923 3 5	219.4	5.43	+.11	625
4 26	277.1	5.22	0	705	5 24	225.5	5.41	.10	630
7 15	283.8	5.19	-.01	717	8 12	231.7	5.40	.09	636
10 3	290.5	5.16	-.02	729	10 31	237.9	5.38	.08	643
12 22	297.3	5.13	-.04	741	1924 1 19	244.1	5.36	+.07	650
1914 3 12	304.2	5.10	-.05	753	4 8	250.4	5.33	.06	659
5 31	311.1	5.08	.06	765	6 27	256.7	5.31	.05	668
8 19	318.1	5.05	.07	776	9 15	263.1	5.28	.04	678
11 7	325.2	5.03	.08	787	12 4	269.6	5.26	.02	689
1915 1 26	332.3	5.01	-.09	796	1925 2 22	276.2	5.23	+.01	700
4 16	339.5	4.99	.10	805	5 13	282.8	5.20	-.01	712
7 5	346.8	4.98	.10	812	8 1	289.5	5.17	-.02	724
9 23	354.1	4.96	.11	817	10 20	296.2	5.14	-.03	736
12 12	1.4	4.96	.11	821	1926 1 8	303.1	5.11	-.05	749
1916 3 1	8.7	4.95	-.11	823	3 29	310.0	5.08	.06	761
5 20	16.0	4.95	.11	823	6 17	317.0	5.06	.07	772
8 8	23.3	4.96	.11	821	9 5	324.0	5.04	.08	783
10 27	30.6	4.96	.11	818	11 24	331.2	5.01	.09	793
1917 1 15	37.9	4.97	-.10	813	1927 2 12	338.4	5.00	-.10	802
4 5	45.2	4.99	.09	806	5 3	345.6	4.98	.10	810
6 24	52.4	5.00	.08	798	7 22	352.8	4.97	.11	816
9 12	59.5	5.02	.07	789	10 10	0.1	4.96	.11	820
12 1	66.6	5.05	.06	778	12 29	7.5	4.95	.11	823
1918 2 19	73.6	5.07	-.05	767	1928 3 18	14.8	4.95	-.11	824
5 10	80.6	5.10	.04	755	6 6	22.1	4.95	.11	823
7 29	87.5	5.12	.02	743	8 25	29.5	4.96	.11	820
10 17	94.3	5.15	-.01	731	11 13	36.8	4.97	.10	815
1919 1 5	101.0	5.18	0	719	1929 2 1	44.0	4.98	-.09	809
3 26	107.7	5.21	+.02	707	4 22	51.2	5.00	.09	801
6 14	114.3	5.24	.03	696	7 11	58.4	5.02	.08	792
9 2	120.8	5.27	.04	684	9 29	65.5	5.04	.07	782
11 21	127.2	5.29	.06	674	12 18	72.6	5.06	.05	771

Jupiter's Co-ordinate. (continued)

Date 12 <sup>h</sup> U.T.	$\lambda_1$	$r_1[\cos\beta_1]$	$r_1\sin\beta_1$	$\frac{10^5}{r_1^3}$	Date 12 <sup>h</sup> U.T.	$\lambda_1$	$r_1[\cos\beta_1]$	$r_1\sin\beta_1$	$\frac{10^5}{r_1^3}$		
1930	3 8	79.6	5.09	-.04	760	1935	1 1	218.6	5.43	+.11	624
	5 27	86.5	5.11	-.03	748		3 22	224.7	5.42	.10	629
	8 15	93.3	5.14	-.01	735		6 10	230.8	5.40	.09	634
	11 3	100.1	5.17	.00	723		8 29	237.0	5.38	.08	641
							11 17	243.3	5.36	.07	648
1931	1 22	106.8	5.20	+.01	711	1936	2 5	249.6	5.34	+.06	657
	4 12	113.4	5.23	.03	699		4 25	255.9	5.31	.05	666
	7 1	119.9	5.26	.04	688		7 14	262.3	5.29	.04	676
	9 19	126.4	5.29	.05	677		10 2	268.8	5.26	.02	687
	12 8	132.8	5.31	.07	667		12 21	275.3	5.23	.01	698
1932	2 26	139.1	5.34	+.08	658	1937	3 11	281.9	5.20	.00	710
	5 16	145.4	5.36	.09	649		5 30	288.6	5.17	-.02	722
	8 4	151.6	5.38	.10	642		8 18	295.3	5.14	-.03	735
	10 23	157.8	5.40	.10	635		11 6	302.1	5.12	-.04	747
1933	1 11	164.0	5.42	+.11	629	1938	1 25	309.0	5.09	-.06	759
	4 1	170.1	5.43	.12	625		4 15	316.0	5.06	.07	771
	6 20	176.2	5.44	.12	621		7 4	323.1	5.04	.08	782
	9 8	182.3	5.45	.12	618		9 22	330.2	5.02	.09	792
	11 27	188.3	5.45	.12	617		12 11	337.4	5.00	.10	802
1934	2 15	194.3	5.46	+.12	616	1939	3 1	344.6	4.98	-.10	809
	5 6	200.4	5.45	.12	616		5 20	351.9	4.97	.11	816
	7 25	206.4	5.45	.12	618		8 8	359.2	4.96	.11	821
	10 13	212.5	5.44	.12	621		10 27	6.5	4.95	.11	824



(in the original size)