# An Experimental Study on the Performances of Drum Type Centrifugal Ventilators of Varying Axial Lengths. 

By

Einosuke Mikumo,

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#### Abstract

. The forms of the runner of the multiblade drum type centrifugal ventilators, of which the "Sirocco" is the best known, and most popularly adopted in our collieries seem somewhat disadvantageous for a uniform flow of air. The air axially sucked in must suddenly change its direction radially in order uniformly to go out from the periphery, but this is impossible unless the air has no moving inertia.

Thus more or less eddy formations due to this cause are inevitable with the runner, and the writer observed that the eddy phenomena decreased by shortening the axial length of the runner. But, then, the ratio, between the amount of frictional surfaces for the air, to the cross rectional area in the casing, would increase, thus causing more frictional loss of energy, if ass.med that the quantity of air treated is proportional to the axial length of the fan.

The purpose of this experiment is to investigate how the fan performences change with fans of varying axial lengths, owing to these two opposite influences, keeping other conditions as unchanged as possible.


## INTRODUCTION.

Of this type of fan of single inlet, it has been suggested, ${ }^{1)}$ when driven without casing, that at the back end the air rushes out while at the inlet end the air is sucked in as shown in fig. I. I have tried to observe the phenomena with a fan of this type which is installed at the Central Laboratory of the Kyoto Imperial University, and have learnt about these unfavorable eddy formations, by bringing a light paper strip close to the periphery of the runner.

[^0]Referring to fig. 1 , the air rushes out considerably at the back end and the velocity seems gradually to diminish to zero up to the middle

Fig. 1.


Fig. 2.
 section, while at the inlet end the paper strip is strongly pushed to the periphery and it appears as if the air were sucked into the fan, and this gradually diminishes up to the middle section.

These eddy phenomena were always observed whatever the number of revolutions. I also tried to observe them, by the method above descrived, at various reduced axial lengths of the fan. Strapping a thick paper ribbon around the inner periphery of the runner as shown in fig. 2 , and increasing the width of the ribbon stepwisely, thus approximately obtaining several fans of different axial lengths, other consitions remaining the same.

Then I observed that the unfavorable phenomena decreased more and more with the fans of shorter axial lengths, and at last the eddy formations apparently ceased and the air was blown out of all parts of the periphery.

Even with fans with casing, I believe that such like phenomena occur.

If it is assumed that the air quantity treated is proportional to the effective axial length of the runner, and consequently to the width of the casing, the velocity of the air at the outlet of the casing should theoretically be a constant, independent of the length of the runner and the width of the casing, other conditions remaining the same. The area of the surfaces along the current, and the cross sectional area in the casing both decrease by shortening the axial length of the fan, but the former does not so rapidly decrease as the latter.

Hence, from the standpoint of the energy losses of the air by surface friction, the fan of longer axial length would be better.

Interested in the question how the fan performances change, owing to these two opposite influences, I have carried out the following experiments.

As the form and number of blades, and the diameter of a fan could not be easily changed, I had to be satisfied with a given fan for my experiments.

## 1. EXPERIMENTAL APPARATUS AND MEASURING INSTRUMENTS.

## § Fan. "Sirocco " type as shown in fig. 3.

Fig. 3.


Runner: Outer diameter
0.251 m

Inner diameter
0.216 m

Axial length (original)
O.III m

Number of blade

Thickness of blade
O.OOI m

Exit angle of blade as shown in the figure $45^{\circ}$
Inlet angle of blade as shown in the figure $1155^{\circ}$
(The above is a model of a Sirocco runner of ordinary proportions.)
Casing: Rectangular cross section.
Profile as shown in fig. 4
Instead of thick paper ribbons strapped around the inner periphery of the runner to change the axial length of the latter as mentioned in the preliminary observations, two short pipes of sheet metal of different diame-

Fig. 4.


Fig. 5.

ters, corresponding to the outer and inner diameters of the runner with a certain amount of clearance, were secured to one side plate of the casing, which is arranged to be movable as ahown in figs. 4 and 5.

By this apparatus the axial width of the casing is also changed at the same time with that of the runner.

As it was difficult to attach evasee of proper side inclinations to the casings of variable widths, the following experiments were proceeded without evasé so: the fan outlets were always opened to the atmosphere.
$\S$ Suction pipe. Circular section as shown in fig. 6.
Inner diameter
0.315 m
Length over all 5.73 m
Length from pipe inlet to the measuring position 3.48 m

The inlet end was made very much like a camera blind and any desired opening is obtained which serve to regulate the air quantity as well as the fan resistance. Without the wire mesh shown in the figure, the meniscus in the manometer oscillated so sensibly that observation was almost impossible, while once the mesh was inserted the oscillation practically ceased so long as the motor speed was constant. Due to this mesh,

the resistance in the suction pipe was, of course, somewhat increased.
$\begin{array}{cc}\S \text { Driving motor. } & \text { Direct Current electric motor as shown in fig. } 7 . \\ \text { Voltage } & \text { IIO } \\ \text { Rated horsed power } & 2 \\ \text { Rated speed } & 1450 \mathrm{rev} / \mathrm{mn}\end{array}$
The motor is one directly coupled to the fan (the latter had been originally fitted with a suitable expanding chimney and a delivery chamber) by the maker, and made as a cradle dynamometer. The stator of the motor is supported by ball bearings.
$\S$ Manometer.
Type——R. Fuess Micromanometer, ${ }^{2 /}$ minimum reading $\frac{1}{100} \mathrm{~mm}$. of alcohol column.
§ Measuring tube.
Type_Prandtl ${ }^{3)}$ as ahown in fig. 8.
§ Revolution counter.
Type-Hasler automatic speed indicator.
§ Psychrometer.
Type——Schubert's "Schleuderpsychrometer ".

## 2. PROCESS OF EXPERIMENT.

In order to carry out the whole series of experiments, the air volume was measured in the suction pipe and the pressure, which the fan generates, was measured also in the suction pipe, for convenience sake.

Regarding the speed of the fan, I have treated it as constant as possible at $1450 \mathrm{rev} / \mathrm{mn}$ throughout all experiments, which is the rated speed of the driving motor. As the currrent for the motor is supplied from the distributing station at the University, the voltage of the supplying main, and consequently the speed of the motor fluctuate within a considerable range, even when the operating condition of the fan is kept the same.

[^1]So, it was necessary to correct all readings, in the following calculations, by using the general rules ${ }^{4}$ that:
when a fan is operating at the same restriction,
i) the volume dilivered zeill vary directly zuith the revolution per minute $N$; and
ii) the various pressires will vary with $N^{2}$; and
iii) the air horse power weill vary weith $N^{3}$. (Accordingly, assuming that the mechanical efficiency remains the same, the shaft horse power weill. also vary with $N^{3}$.)

The rheostat was regulated, when the speed fluctuation had surpas$\mathrm{sed} \pm 50 \mathrm{rev} / \mathrm{mn}$.

The air volume was measured with the measuring tube inserted into the suction pipe. Equally dividing the cross sectional area at the

4) Lionel S. Marks, "Mech. Eng. Handbook", 1545 (1916)
measuring section by 8 concentric areas and in each ring the velocity heads were observed at two points which lie symmetrically, to the center, on a diameter as shown in fig. 9 .

To take the center velocity into account, velocity heads at three points were observed, in the center circle. Thus, referring to the figure, if the letters $1,2,3,4, \ldots \ldots . C \ldots .4^{\prime}, 3^{\prime}, 2^{\prime}, 1^{\prime}$ represent the velocities of the air at the points above mentioned, then

$$
\begin{aligned}
& \text { Volume }= \frac{\pi d^{2}}{4 \times 8}\left(\frac{\mathrm{I}+\mathrm{1}^{\prime}}{2}\right)+\frac{\pi d^{2}}{4 \times 8}\left(\frac{2+2^{\prime}}{2}\right)+\ldots \ldots \ldots . \\
& \quad+\frac{\pi d^{2}}{4 \times 8}\left(\frac{7+7^{\prime}}{2}\right)+\frac{\pi d^{2}}{4 \times 8}\left(\frac{8+C+8^{\prime}}{3}\right) \\
&= \frac{\pi d^{2}}{4}\left(\frac{\mathrm{I}+2+3+\ldots \ldots+7+7^{\prime}+6^{\prime}+\ldots \ldots+\mathrm{I}^{\prime}}{2 \times 8} .\right. \\
&\left.\quad+\frac{8+C+8^{\prime}}{3 \times 8}\right) \ldots \ldots \ldots \ldots(\mathrm{I})
\end{aligned}
$$

where $d$ is the diameter of the suction pipe at the measuring section $=$ 0.315 m . By the way, the term in the bracket represent the mean volocity.

The suction inlet opening was regulated at seven stages, namely, $o$, 7, 11, 14, 17, 20, and 30 cm diameters which were determined so in the following way; driving the fan, the effective axial length of the runner of which is o.IIIm $\left(=4 \frac{3^{\prime \prime}}{8}\right)$ at the speed above mentioned, and regulating the inlet opening at seven stages with trials, so as to approximately obtain equal amounts of increase of air volume. To measure the volume then, an anemometer was used at the discharge end of the fan.

The diameters of the suction inlet opening above mentioned were kept the same at other axial lengths of the fan.

To avoid confusion, let them be symbolized in the experiments as follows;

| Suction inlet diametres | $0^{\mathrm{cm}}$ | $7^{\mathrm{cm}}$ | ${ }^{11}$ | $14^{\mathrm{cm}}$ | $17^{\mathrm{cm}}$ | $20^{\mathrm{cm}}$ | $30^{\mathrm{cm}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Symbols | 0 | I | II | 1 II | IV | V | VI |

and for the axial lengths of the runner, and consequently the widths of the casing (refer to fig. 4.), let them be denoted as follows;

| Width of casing outlet in | $4 \frac{3}{4}$ | 4 | $3-\frac{1}{4}$ | $2 \frac{1}{2}$ | 1-3 ${ }^{4}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | 12.1 | 10.1 | 8.3 | 6.4 | 4.4 |
| cffective axial lengthof runner in $\left\{\begin{array}{c}\text { inch } \\ \mathrm{cm}\end{array}\right.$ | $4 \frac{3}{8}$ | $3 \frac{5}{8}$ | $2-\frac{7}{8}$ | $2-\frac{1}{8}$ | $1 \frac{3}{8}$ |
|  | 11.1 | 9.2 | 7.3 | $5 \cdot 4$ | 3.5 |
| Symbols. | 1 | 2 | 3 | 4 | 5 |

Thus, for example, by denoting the experiment with 4 at II, the effective axial length of the runner, casing outlet width, and the suction inlet opening correspond to $5.4 \mathrm{~cm}, 6.4 \mathrm{~cm}$, and 11 cm respectively.

Throughout my experiments, the temperature of the air in the suction pipe as well as that at delivery end were practically equal to the atmospheric temperature; the differences between these three were so small that I could not detect the differences, with ordinary Centigrade thermometers, so in all cases in my experiments these three temperatures were assumed to be the same.

As the density of the air, though in small range, varies with the sucsion resistance, and in pretty large range, with barometric conditions, atmospheric temperature and humidity, the volume obtained were reduced, in drawing the characteristic curves, to a standard condition ${ }^{5)}$ of 735.5 mm Hg and $15^{\circ} \mathrm{C}$ for convenience in comparing the observed results.

## 3. CALCULATION EXAMPLES.

The formulae, employed for the calculations in my experiments, are summarized as follows, with corresponding numerical examples for the case of 1 at II.

## § BAROMETER CORRECTION.

$$
\begin{equation*}
b=b^{\prime}(\mathrm{I}-0.000 \mathrm{I} 62 \times t)^{6)} \tag{2}
\end{equation*}
$$

where $\quad b=$ corrected barometric height in mm Hg .

[^2]\[

$$
\begin{aligned}
& b^{\prime}=\text { barometer reading (the scale is of brass) } \\
& t=\text { temperature } \text { in }^{\circ} \mathrm{C} \\
& \text { Example } 1 . \\
& t=6^{\circ} \mathrm{C} \\
& b^{\prime}=765 \\
& \therefore \quad b=764.3 \mathrm{~mm} \mathrm{Hg} \\
& \text { absolute } \text { pressure }=13.596 \times 764.3=1039 \mathrm{I} .42 \mathrm{~kg} / \mathrm{m}^{2} \\
& \text { or } \quad 1039 \mathrm{I} .42 \mathrm{mmAq} \quad\left(\because \quad \mathrm{I} \mathrm{mmAq}=\mathrm{Ikg} / \mathrm{m}^{2}\right)
\end{aligned}
$$
\]

Note Second decimal figure corresponds to the minimum reading of the micromanometer.
§ HUMIDITY IN ATMOSPHERE.

$$
\begin{equation*}
\varphi_{0} h_{0}=h_{f}-k b\left(t_{0}-t_{f}\right)^{7)} \tag{3}
\end{equation*}
$$

where $\quad \varphi_{0}=$ relative humidity.
$t_{0}=$ temperature in ${ }^{\circ} \mathrm{C}$. given by dry thermometer.
$t_{f}=$ ditto given by wet thermometer.
$h_{0}=$ water vapour tension at $t_{0}$ in $\mathrm{mm} \mathrm{Hg}{ }^{8)}$
$h_{f}=$ ditto at $t_{f}$
$b=$ as before.
$k=\mathrm{a}$ constant, taking 0.0007 following the consideration in Watson's text, ${ }^{\text {T }}$
Example 2.

$$
t_{0}=6^{\circ} \mathrm{C} \quad t_{f}=4^{\circ} \mathrm{C}
$$

$$
\varphi_{0}=\frac{6.1-0.0007 \times 764.3(6-4)}{7.0}=0.72
$$

## § DENSITY OF MANOMETER LIQUID.

Alcohol is used as manometer liquid, and its density was measured with a hydrometer as 0.8265 at. $15^{\circ} \mathrm{C}$, and the densities at other temperatures may be given by the formula below ;

$$
\begin{equation*}
r_{t}=\frac{\mathrm{I}}{[\mathrm{I}+a(t-\mathrm{I} 5)]} \gamma_{15} \tag{4}
\end{equation*}
$$

where $\quad t=$ temperature in ${ }^{\circ} \mathrm{C}$.

[^3]Example 3.

$$
\begin{aligned}
\gamma_{1} & =\text { density of the alcohol at } t^{\circ} \mathrm{C} . \text { in } \mathrm{g} / \mathrm{cm}^{3} \\
\gamma_{15} & =\text { ditto at } 15^{\circ} \mathrm{C}=0.8265 \\
\alpha & =\text { expansion coefficient of alcohol, taking } 0.001 \mathrm{I}^{9)}
\end{aligned}
$$

$$
t=6^{\circ} \mathrm{C}
$$

$$
r_{t}=0.8265 /[1+0.001 \mathrm{I}(6-15)]=0.835
$$

## § PRESSURE DEPRESSION IN SUCTION PIPE. D.

Connecting the side opening of the measuring tube to the manometer, the static depressicn is obtained. As the difference in reading was not noticeable by placing the tube at the various positions in the measuring section, the tube was always inserted at the center of the pipe for the purpose.
Example 4. Manometer inclination $=15^{\circ}$
density of alcohol $=0.835$ at $6^{\circ} \mathrm{C} \quad$ reading $=164 \mathrm{~mm}$ speed of the fan $=1450 \mathrm{rev} / \mathrm{mn}$ (exactly)
static depression $D=164 \times \sin \mathrm{I} 5^{\circ} \times 0.835=35.44$
mmAq or $\mathrm{kg} / \mathrm{m}^{2}$
§ HUMIDITY OF AIR IN SUCTION PIPE.
Let $\varphi_{s}=$ relative humidity of air in suction pipe.
$\varphi_{0}=$ ditto of atmospheric air as before,
$h_{0}=$ water vapour tension at atmospheric temperature in $\mathrm{kg} / \mathrm{m}^{2}$,
$h_{s}=$ ditto at the temperature of air in suction pipe,
$p_{0}=$ atmospheric pressure in $\mathrm{kg} / \mathrm{m}^{2}$,
$p_{s}=$ absolute static pressure in suction pipe at measuring point in $\mathrm{kg} / \mathrm{m}^{2}$,
then

$$
\varphi_{s} \frac{h_{s}}{p_{s}}=\varphi_{0} \frac{h_{0}}{p_{0}}{ }^{10)}
$$

but by the assumptions regarding the temperatures above described, $h_{0}=h$

[^4]10) "Hütte" 20 Auf. I. 322
and the formula becomes much simpler so that,
\[

$$
\begin{equation*}
\varphi_{s}=\varphi_{0} \frac{p_{s}}{p_{0}} . \tag{5}
\end{equation*}
$$

\]

Example 5. $\quad \varphi_{0}=0.72 \quad p_{0}=1039 \mathrm{I} .42 \mathrm{~kg} / \mathrm{m}^{2}$

$$
\begin{aligned}
p_{s} & =1039 \mathrm{I} .42-35.44=10355.98 \mathrm{~kg} / \mathrm{m}^{2} \\
\varphi_{s} & =0.72(\mathrm{IO} 355.98 / \mathrm{IO} 39 \mathrm{I} .42)=0.72
\end{aligned}
$$

As seen in this example, the influence of the pressure drop on the humidity is negligible in my cases.

## § AIR CONSTANTS.

Let $\quad R_{0}=$ air constant for atmospheric air,
$R=$ air constant for dry air $=29.27$
$p_{0}, h_{0}$ and $\varphi_{0}=$ as before

> then,

$$
R_{0}={\frac{R}{\mathrm{I}-0.377 \varphi_{0} \frac{h_{0}}{p_{0}}}}^{11)}
$$

Example 6. $\varphi_{0}=0.72 \quad p_{0}=10391.42 \mathrm{~kg} / \mathrm{m}^{2}$

$$
h_{0}=13.596 \times 7\left(=\text { tension at } 6^{\circ} \mathrm{C} \text { in } \mathrm{mm} \mathrm{Hg}\right)
$$

$$
=95,2 \mathrm{~kg} / \mathrm{m}^{2}
$$

$$
R_{0}=\frac{29.27}{\mathrm{I}-0.377 \times 0.72 \frac{95.2}{\mathrm{I} 039 \mathrm{I} .4^{2}}}=29.34
$$

Similarly let $R_{s}=$ air constant for the air in suction pipe, $\varphi_{s}, h_{s} . p_{s}$ and $R=$ as before, then

$$
\begin{equation*}
R_{s}=\frac{R}{\mathrm{I}-0.377 \varphi_{\mathrm{s}} \frac{h_{s}}{p_{s}}} \tag{7}
\end{equation*}
$$

Example 7. $\quad \varphi_{s}=0.72 \quad p_{s}=10355.98 \mathrm{~kg} / \mathrm{m}^{2} \quad h_{s}=h_{0}=95.2 \mathrm{~kg} / \mathrm{m}^{2}$

$$
R_{s}=\frac{29.27}{\mathrm{I}-0.377 \times 0.72 \frac{95.2}{10355.98}}=29.34
$$

As in these examples, $R_{0}$ is practically equal to $R_{s}$ throughout my experiments.
II) Ostertag, "Kolben-und 'Turhokompressoren" ro (1919)

## § DENSITY OF AIR.

| Atmospheric air. | $\begin{equation*} \gamma_{0}=\frac{p_{0}}{R_{0} T_{0}} \tag{8} \end{equation*}$ |
| :---: | :---: |
| where $\quad \gamma_{0}=$ specific weight of air in $\mathrm{kg} / \mathrm{m}^{3}$ |  |
| $R_{0}$ and $p_{0}=$ as before |  |
| $T_{0}=$ absolute temperature of air in ${ }^{\circ} \mathrm{C}$. |  |
| Example 8. | $\begin{aligned} & T_{0}=273+6=279^{\circ} \mathrm{C} \quad p_{0}=10391.42 \mathrm{~kg} / \mathrm{m}^{2} \\ & R_{0}=29.34 \quad \text { then } \gamma_{0}=1.269 \mathrm{~kg} / \mathrm{m}^{3} \end{aligned}$ |
| Air in suction pipe | $\gamma_{s}=\frac{p_{s}}{R_{s} T_{s}}$ |
| where $\quad \gamma_{s}=$ specific weight of air in suction pipe in $\mathrm{kg} / \mathrm{m}^{3}$ |  |
| Example 9. | $=$ absolute temperature ${ }^{\circ} \mathrm{C}$ in suction pipe. |
|  | $T_{s}=T_{0}=279^{\circ} \mathrm{C} \quad p_{s}=10355.98 \mathrm{~kg} / \mathrm{m}^{3}$ |
|  | $R_{s}=R_{0}=29.34$, then $r_{s}=\mathrm{I} .265 \mathrm{~kg} / \mathrm{m}^{3}$. |

Air at delivery end. The density of air at the delivery end is assumed to be equal to that of atmospheric air.

## § AIR VELOCITY IN SUCTION PIPE.

Let $\quad h_{v}=$ velocity head in $\mathrm{mmAq} . \quad v_{s}=$ air velocity in $\mathrm{m} / \mathrm{s}$
$g=$ gravity acceleration $=9.80 \mathrm{~m} / \mathrm{s}^{2}$
$\gamma_{s}=$ density of air in $\mathrm{kg} / \mathrm{m}^{3}$
$\boldsymbol{\xi}=$ measuring tube constant, always assuming $\xi=\mathrm{I}$,
then, modifying the general formula $h_{v}=\xi \gamma_{s} v_{s}^{2} / 2 \mathrm{~g}$, we obtain

$$
\begin{equation*}
v_{s}=\sqrt{\frac{2 g}{\gamma_{s}}} \sqrt{h_{v}} \tag{io}
\end{equation*}
$$

Example ro. At the pipe center, the reading of velocity head was 26 mm alcohol, inclination of the manometer $=\mathrm{I}^{\circ} \mathrm{IO}^{\prime}$, density of alcohol $=0.835$ $\mathrm{g} / \mathrm{cm}^{3}, \quad r_{s}=1.265 \mathrm{~kg} / \mathrm{m}^{3}$, speed of the fan $=1440 \mathrm{rev} / \mathrm{mn}$. then, $h_{v}=26 \times \sin 1^{\circ} \mathrm{I} O \times 0835=0.435 \mathrm{mmAq}$.

$$
h_{v} \text { reduced to } 1450 \mathrm{rev} / \mathrm{mn}=0.435 \times \frac{145 \mathrm{o}^{2}}{\mathrm{I} 44 \mathrm{O}^{2}}=0.44 \mathrm{r}
$$

$$
\tau_{s}=\sqrt{\frac{2 \times 9.80}{1.265}} \cdot \sqrt{0.44 \mathrm{I}}=2.62 \mathrm{~m} / \mathrm{s}
$$

§ VOLUME REDUCED TO ( $735.5 \mathrm{mmHg}, \mathrm{I} 5^{\circ} \mathrm{C}$ )

$$
V_{r}=V_{s} \frac{273+15}{273+t} \cdot{\frac{p_{s}}{7.35 .5 \times 13.596}}^{197}
$$

where $\quad V_{r}=$ reduced volume $V_{s}=$ volume measured $t=$ air temperature in suction pipe in ${ }^{\circ} \mathrm{C}$. $p_{s}=$ as before.
Example ir. $\quad V_{s}=0.1948 \mathrm{~m}^{3} / \mathrm{s}$ (refer to Table 41)
$t=6^{\circ} \mathrm{C}$
$\therefore \quad V_{r}=0.208 \mathrm{~m}^{3} / \mathrm{s}$
§ THEORETICAL ISOTHERMAL POWER, E and E'.
Power E in P. S. $=p_{1} V_{g} \ln \frac{p_{2}}{p_{1}} / 75$ $\qquad$
where $\quad p_{1}=$ absolute suction pressure in $\mathrm{kg} / \mathrm{m}^{3}$
$=$ absolute static suction pressure + suction velocity pressure.

$$
\begin{aligned}
p_{2}= & \text { absolute delivery pressure in } \mathrm{kg} / \mathrm{m}^{3} \\
= & \text { absolute static delivery pressure }+ \text { delivery velocity } \\
& \text { pressure. }
\end{aligned}
$$

When the effective isothermal power $\mathrm{E}^{\prime}$ is required, $p_{2}$ is put equal to the absolute static delivery pressure, ${ }^{13)}$ that is, in my case, it is equal to the atmospheric pressure $p_{0}$, as the delivery velocity pressure may be assumed to be totally lost.
Example 12. Let $p_{0}=1039 \mathrm{I} .42 \mathrm{~kg} / \mathrm{m}^{2}, \quad p_{s}=10355.98 \mathrm{~kg} / \mathrm{m}^{2}$, mean velocity in suction pipe $=2.44 \mathrm{~m} / \mathrm{s}$ (refer to Table $4_{1}$ ), outlet area of fan casing $=0.0231 \mathrm{~m}^{2}$.
12) Gramberg, "Tech. messungen " 112
13) Gramberg, "Masch. untersuchungen" 471 (192I)

Then, suction velocity pressure $=\frac{(2.44)^{2}}{2 \times 9.80} \times 1.265=0.38 \mathrm{~kg} / \mathrm{m}^{2}$.
$\therefore \quad p_{1}=10355.98+0.38=10356.36 \mathrm{~kg} / \mathrm{m}^{2}$.
The volume delivered at discharge end
$=V_{s} \times \frac{\gamma_{s}}{\gamma_{0}}=0.1948 \times 1.265 / \mathrm{I}, 269=0.1942 \mathrm{~m}^{3} / \mathrm{s}$, so the
mean delivery volocity $=0.1942 / 0.023 \mathrm{I}=8.4 \mathrm{I} \mathrm{m} / \mathrm{s}$, then,
delivery velocity pressure $=\left(8.4 \mathrm{I}^{2} / 2 \times 9.80\right) \times \mathrm{I} .269=4.58 \mathrm{~kg} / \mathrm{m}^{2}$

$$
\therefore \quad p_{2}=10391.42+4.58=10396.00 \mathrm{~kg} / \mathrm{m}^{2}
$$

or when the effective power is required,

$$
p_{2}=10391.42 \mathrm{~kg} / \mathrm{m}^{2} .
$$

Total power E in P.S. $=10356.36 \times 0.1948 \ln \frac{10396.00}{10356.36} / 75=0.102$
Effective power $\mathrm{E}^{\prime}$ in P. S. $=0.09 \mathrm{I}$
§ THEORETICAL ADIABATIC POWER.
Power in P.S. $=p_{1} V_{3} \frac{x}{x-1}\left[\left(\frac{p^{2}}{p_{1}}\right)^{\frac{x-1}{x}}-1\right] / 75$
where

$$
\begin{equation*}
x=\frac{C_{p}^{\prime}+m C_{p}^{\prime \prime}}{C_{v}^{\prime}+m C_{v}^{\prime \prime}}{ }^{14)} \tag{13}
\end{equation*}
$$

and $\quad C_{p}^{\prime}=$ specific heat of air at constant pressure $=0.2375$
$C^{\prime}{ }_{v}=$ ditto at constant volume $=0.1685$
$C^{\prime \prime}{ }_{p}=$ specidic heat of superheated steam at constant pressure $=0.4805$
$C^{\prime \prime}{ }_{v}=$ ditto at constant volume $=0.3695$
and

$$
m=0.623 p_{d} / p_{2}
$$

where $\quad p_{d}=$ vapour tention at a temperature in $\mathrm{kg} / \mathrm{m}^{2}$ $p_{i}=$ (partial) pressure of dry air in $\mathrm{kg} / \mathrm{m}^{2}$ at the same temperature.
Example 13. All data are as in the last example.
$p_{l}=p_{s}-p_{d}=p_{s}-13.596 \times$ vapour tention at $6^{\circ} \mathrm{C}$ in mm $\mathrm{Hg} \times$ relative humidity $\varphi_{s}$

[^5]\[

$$
\begin{aligned}
& =10355.08-\mathrm{I} 3.596 \times 7 \times 0.72=10355.98-68.5 \\
& =10287.48 \mathrm{~kg} / \mathrm{m}^{2} \\
m & =0.623 \times 68.5 / \mathrm{I} 0287.48=0.00414 \\
x & =\frac{0.2375+0004 \mathrm{I} 4 \times 0.4805}{0.1685+0.004 \mathrm{I} 4 \times 0.3695}=1.409 \\
& x \\
x-\mathrm{I} & =3.445 \quad \frac{x-1}{x}=0.2903
\end{aligned}
$$
\]

Power in P. S. $=10356.36 \times 0.1948$

$$
\times 3.445\left[\left(\frac{10396}{10356.36}\right)^{0.2203}-1\right] / 75=0.102
$$

As in these examples, the isothermal and adiabatic powers are practically equal to each other in such small ranges of pressure differences as in my cases, hence for simplicity, the theoritical horse powers are calculated always as isothermal.

## § PRESSURE DIFFERENCES BETWEEN SUCTION AND DELIVERY, P and $\mathrm{P}^{\prime}$.

Although the pressure difference can be obtained as ( $p_{2}-p_{1}$ ) using the above described figures, but to avoid such large numerical values as (ro396.00-10356.36) it may be more convenient to calculate as follows:

As $p_{2}=$ Atmospheric pressure + static delivery pressure + delivery velocity pressure, and
$p_{1}=$ atmospheric pressure-static suction depression + suction velocity pressure.
$p_{2}-p_{1}=$ static delivery pressure + delivery velocity pressure + static suction depression-suction velocity pressure.

## Example 14.

As static delivery pressure (over atmospheric) has assumed to be zero, the total pressure difference $P=p_{2}-p_{1}=4.58+35.44-0.38=39.64$ $\mathrm{kg} / \mathrm{m}^{2}$ or mmAq , and the effective pressure difference $\Gamma^{\prime}=p_{2}-p_{1}=35.44$ $-0.38=35.06 \mathrm{~kg} / \mathrm{m}^{2}$

Note: If an ideal frictionless evasee were attached to the casing, it might be possible, (by reducing the inlet area of the suction pipe) to
treat an equal amount of air which the fan without evasée has treated, against certain higher ventilating resistance (the driving power remaining the same). In such a condition the delivery velocity head which had been otherwise lost would converge towards null and serve to create more depression in the suction pipe if necessary; and in the ideal limit, the total pressure difference obtained as in the above example would all be utilized in the effective suction depression. Thus the total pressure difference is the ideal maximum possible limit of effective pressure difference with evasée.

## § THEORETICAL HEAD WHICH THE FAN GENERATES.

The theoretical maximum possible pressure difference between suction and delivery is obtain by the formula:

$$
H=\frac{u^{2}}{g}+\frac{u}{g} C \cdot \cot \beta^{15)}
$$

$\qquad$
which represents the best case and the formula is called that for " Normal" condition,
where $\quad H=$ head the fan creates in $m$ air column.
$u=$ outer peripheral velocity in $\mathrm{m} / \mathrm{s}$.
$C=$ radial exit velocity in $\mathrm{m} / \mathrm{s}$
$\beta=$ outer vane-angle as shown in fig. 3 .
$g=$ gravity acceleration $=9.80 \mathrm{~m} / \mathrm{s}^{2}$.
Example 15. Suction air volume $=0.1948 \mathrm{~m}^{3} / \mathrm{s}$ (reduced to the fan speed of $1450 \mathrm{rev} / \mathrm{mn}$ ),
outer diameter $=0.251 \mathrm{~m}$,
$\beta=45^{\circ} \quad \therefore \quad \cot \beta=\mathrm{I}$,
number of blades $=56$,
thickness of blades $=0.001 \mathrm{~m}$,
axial length of blade $=0.111 \mathrm{~m}$.
Then the net area of outflow at the outer periphery is

$$
\pi \times 0.25 \mathrm{I} \times 0.11 \mathrm{I} \times\left[1-0.001 \times 56 /\left(\pi \times 0.25 \mathrm{I} \times \sin 45^{\circ}\right)\right]
$$

[^6]$$
=0.0876 \times 0.9 \mathrm{~m}^{2}
$$
and assuming the coefficient of contraction of flow to be 0.92 following Ostertag, and moreover assuming that it is constant at various axial lengths of blades, then the effective exit area of flow becomes,
\[

$$
\begin{aligned}
& 0.0876 \times 0.9 \times 0.92=0.0725 \mathrm{~m} \\
& C=\left(\text { suction air volume } \times \gamma_{s} / \gamma_{0}\right) \div 0.0725 \\
& \quad=(0.1948 \times 1.265 / \mathrm{I} .269) \div 0.0725=2.678 \mathrm{~m} / \mathrm{s}
\end{aligned}
$$
\]

Strictly speaking, the volume change must not be referred to $\gamma_{0}$, but to the density of air at the exit of the runner. But in my case, as the casing directly opens in the atmosphere, it may be allowed to do so by approximation.

$$
\begin{aligned}
& u=(\pi \times 0.25 \mathrm{I} \times 1450) \div 60=19.06 \mathrm{~m} / \mathrm{s} \\
& u^{2} / g=37.07 \quad \mathrm{u} / \mathrm{g}=1.945 \\
& H=37.07+\mathrm{r} .945 \times 2.678=42.27 \mathrm{~m} \text { air column. }
\end{aligned}
$$

As the fan has created 42.27 m of head, treating the atmospheric air whose density is $1.269 \mathrm{~kg} / \mathrm{m}^{3}$; converting which to the pressure difference $\mathrm{H}_{w}$, we obtain

$$
H_{w}=\mathrm{I} .269 \times 42.27=53.64 \mathrm{~kg} / \mathrm{m}^{2} \text { or } \mathrm{mmAq}
$$

## § FUN INPUT POWER Ef AND ITS MEASUREMENT.

$$
\begin{equation*}
E_{f}=\text { Power in P. S. }=\mathrm{M}_{\iota} \omega / 75 \tag{15}
\end{equation*}
$$

where $\quad M_{c}=$ moment for net output of motor in mkg.
$\omega=$ angular velocity in rad/s
As the fan is driven by a motor which was made as a cradle dynamometer as ahown in figs. 6 and 7 ,

$$
M_{e}=M_{a}-\left(V+R_{1}+R_{2}\right)^{16)}
$$

where $\quad V=$ moment for motor windage.
$R_{1}=$ moment for motor shaft friction.
$R_{2}=$ moment for the ball bearing friction of the stator.
$M_{d}=$ moment measured in mkg.
As it is difficult to measure the value $V, R_{1}$, and $R_{2}$ individually,

[^7]the aggregate values was measured by driving the motor with no load at $1450 \mathrm{rev} / \mathrm{mn}$, that is, without the fan, and the value of momont was 15 grammes weight on the pan at Im leverage from the motor center; moreover, the moment were practically constant from a speed of 1400 up to $1500 \mathrm{rev} / \mathrm{mn}$.

To avoid confusion, the leverage was kept constant at 1 m , and a weight of 15 g was secured to the pan during the whole experiment. Then the weight newly put on corresponds to $M_{e}$.
Example 16. (additional) weight $=78 \mathrm{~g}$.
motor speed $=1455 \mathrm{rev} / \mathrm{mn}$
then, $\quad M_{e}=0.078 \times \mathrm{I}=0.078 \mathrm{mkg}$.
$\omega=2 \times 1455 / 60=152.33 \mathrm{rad} / \mathrm{s}$
Power in P.S. $=0.078 \times 152.33 / 75=0.159$
$E_{f}=$ Power reduced to $1450=0.159 \times 1450^{3} / 1455^{3}$
$=0.158$
§ MECHANICAL EFFICIENCIES.
Let $\quad \eta_{m}=$ total mechanical effiency, and $\eta^{\prime}{ }_{m}=$ effective mechanical efficiency, then
$\eta_{m}=\frac{\text { theoretical total isothermal power } E}{\text { fan input power }} \ldots \ldots$.....( 6
$\gamma^{\prime}{ }_{m}=\frac{\text { theoretical effective isothermal power } E^{\prime}}{\text { fan input power }} \ldots$ (17)
Example 17.

$$
\begin{aligned}
\eta_{m} & =\frac{0.102}{0.158}=0.652 \\
\eta_{m}^{\prime} & =\frac{0.091}{0.158}=0.575
\end{aligned}
$$

## § PRESSURE EFFICIENCIES.

Let $\quad \eta_{p}=$ total pressure efficiency, and
$\eta_{p}^{\prime}=$ effective pressure efficiency, then

$$
\begin{align*}
\eta_{p} & =\frac{\text { observed total pressure difference } P}{-(\text { I } 4) \times \gamma_{0}} \ldots  \tag{I8}\\
\eta_{p}^{\prime} & =\frac{\text { observed effective pressure difference } P^{\prime}}{(\mathrm{I} 4) \times \gamma_{0}} \tag{I9}
\end{align*}
$$

Example 18.

$$
\begin{aligned}
& \eta_{p}=\frac{39.64}{53.64}=0.74 \\
& \eta_{p}^{\prime}=\frac{3506}{53.54}=0.653
\end{aligned}
$$

## § VOLUMETRIC CAPACITY.

It is the ratio of the suction air volume treated, during one revolu. tion, to the cubical content of the runner. ${ }^{17)}$

$$
\begin{equation*}
\eta_{v}=V_{0} / \frac{\pi d^{2}}{4} . \text { b. n. } \tag{20}
\end{equation*}
$$

where $\quad \gamma_{v}=$ volumetric capacity.
$d=$ outer diameter of the runner.
$b=$ axial length of the runner.
$n=$ speed of the fan in rev $/ \mathrm{s}$.

## Example 19.

$$
\begin{aligned}
& d=0.25 \mathrm{Im}, \quad b=0.11 \mathrm{Im} \quad V_{s}=0.1948 \mathrm{~m}^{3} / \mathrm{s} \quad n=.1450 / 60 \\
& \text { then } \quad \eta_{v}=1.45
\end{aligned}
$$

## 4. SUMMARY OF EXPERIMENTAL AND CALCULATED DATA.

Table I shows the atmospheric conditions, etc.
Tables 2 and 3 show the static pressures, humidities, and densities of air in the suction pipe.
Tables $4_{1}, 42,48,44$ and 45 show the air velocities and volumes in the suction pipe.
17) Lionel S. Marks, "Mcch. Eng. Handbook." 1542.

Table 5 shows velocity pressures of suction and delivery, and the theoretical isothermal powers.
Table 6 shows pressure differences between suction and delivery.
Table 7 shows fan input powers.
Table 8 shows pressure efficiencies, mechanical efficiencies and volumetric capacities.
Plates $1,2,3,4$, and 5 show the performances of the fans $1,2,3,4$, and 5 .
Plate 6 is the summary of the curves of the static depressions $D$ and those of the effective pressure differences $P$.
Plate 7 is the summary of the curves of the effective isothermae powers $E^{\prime}$.
Plate 8 is the summary of the curves of the total isothermal powers $E$ and those of the fan input powers $E_{f}$.
Plate $\quad 9$ is the summary of the curves of total pressure differences $P$.
Plate . Io is the summary of the curves of the theoretical head $H_{w}$.
Plate II is the summary of the curves of the total mechanical efficiencies $\eta_{m}$.
Plate 12 is the summary of the curves of the effective mechanical efficiencies $\eta_{m}^{\prime}$.
Plate 13 is the summary of the curves of the total pressure efficiencies $\eta_{p}$.
Plate 14 is the summary of the curves of the effective pressure efficiencies $\eta^{\prime}{ }_{p}$.
Plate 15 is the summary of the curves of the volumetric capacities $\eta_{v}$.

## 5. CONCLUSIONS.

Although my experiments were carried out under the conditions of variable barometric heights, atmospheric temperatures and humidities, and the various data are not reduced to certain standards, excepting that the characteristic curves are plotted on the reduced volumes as abscissae, yet the general tendencies of performances of the fans of variable axial lengths
without evasée might be indicated.
By shortening the axial length of the fan (whose dimensions, forms of blades, and the profile of the casing are shown in figs. 3 and 4), and on the other hand, regulating the inlet area of the suction pipe and drivlng the fan at the constant speed of $1450 \mathrm{rev} / \mathrm{mn}$, the following conclusions result so far as my experiments are concerned:

1 , In a certain range of suction restrictions, for instance, between those of II and III, it is noticeable that the total pressure differences (plate 9), the total pressure efficiencies (Pl. I3), and the mechanical efficiencies (PL. II and I2), of the fans of shorter axial lengths are higher, in a certain extent, than those of the fans of longer axial lengths.
2. But in the majority of suction restrictions, the above mentioned characteristics are better in the fan of longer axial length.
3. As regards the effective pressure efficiencies (PI. 14), the fan of maximum axial length always shows the best results and the longer the length the better become these two.
4. As regards the theoretical heads (PI. IO) and the volumetric capacities (pl. I5), the fan of shorter axial lengths show better results in the majority of suction restrictions. These may indicate that the eddy formations of air at the runner, before stated, gradually decrease more and more with the fans of shorter axial lengths.

Combining these items and taking the other characteristics into consideration, it may be said further that:
5. If the eddy formations, before stated, decrease and consequently the loss of energy by this cause decrease, by reducing the length of the fan, the surface frictional resistance in the casing, on the contrary, might have the predominating influence on the internal resistance of the fan, in general.
6. If a fan of this type has to be operated in a wide range of suction restrictions, that of maximum length gives the best results in general aspects, and consirdering the tendencies shown in the experiments, it might be allowed to conjecture that, regarding fan performances, a fan of this type of longer axial length than ordinary is, to a certain extent,
better for the purpose.

May 25 th, 1924.
Mining Laboratory •
Kyoto Imperial University'

Table 1. Atmospheric Conditions etc.

| Fan | 1 | 2 | 3 | 4 | 5 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Date | 1-eb. 5 | Feb. 7 | Feb. 9 | Feb. 20 | Feb. 23 | 1924 |
| Duration of test | 1-4 P.M. | 2.5 P.M. | 1-4 P.M. | 1-4 P.M. | 9-12 P.M. |  |
| Weather | cloudy | rainy | cloudy | cloudy | fine |  |
| Barometer reading | 765.0 | 758.4 | 760.8 | 756.5 | 754.0 | mm Hg |
| Ditto corrected by (2) | 764.3 | 757.5 | 759.8 | 755.6 | 753.4 |  |
| dry thermometer | 6 | 7.5 | 8 | 8 | 5 | $\mathrm{C}^{\circ}$ |
| wet thermometer | 4 | 6 | 5 | 5 | 2 |  |
| Ralative humidity by (3) . | 0.72 | 0.80 | 0.61 | 0.61 | 0.57 |  |
| Air constant by (6) . . . . . . | 29.34 | 29.36 | 29.34 | 29.34 | 29.33 |  |
| Density of air by (8). | 1.269 | 1.251 | 1.253 | 1.246 | 1.222 | $\mathrm{kg} / \mathrm{m}^{3}$ |
| Density of alcohol by (4) . . | 0.835 | 0.834 | 0.833 | 0.833 | 0.836 | $\mathrm{g} / \mathrm{cm}^{3}$ |

Table 2. Static pressures in suction pipe.

| $\underset{\text { resistance }}{\text { Fan }}$ | Fan speed in rev/mu |  |  |  |  | Reading of static depression in mm of alcohol, by manometer inclination $=15^{\circ}$ |  |  |  |  | Static depression D reduced to $145^{\circ}$ rev/mn in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$ |  |  |  |  | Absolute pressure in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$ |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 1 | 2 | 3 | 4 | 5 | 1 | 2 | $\boldsymbol{3}$ | 4 | 5 | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 |
| $\bigcirc$ | 1430 | 1440 | 1450 | 1460 | 1460 | 174 | 157 | 145 | 130 | 103 | 38.68 | '34.36 | 31.26 | 27.60 | 22.00 | 10352.74 | 10264.61 | 10298.98 | 10245.54 | 10221.23 |
| I | 1450 | 20 | 20 | 30 | 50 | 169 | 147 | 134 | 124 | 105 | 36.52 | 33.06 | 30.11 | 27.50 | 22.72 | 10354.90 | 265.91 | 300.13 | 245.64 | 220.51 |
| II | 1450 | 20 | 20 | 30 | 25 | 164 | 155 | 145 | 118 | 76 | 35.44 | 34.90 | 32.58 | 25.10 | 17.00 | 10355.98 | 264.07 | 297.66 | 248.04 | 226.23 |
| III | 1440 | 35 | 75 | 20 | 15 | 165 | 149 | 131 | 89 | 46 | 36.09 | 32.85 | 27.29 | 20.00 | 10.45 | 10355.33 | 266.12 | 302.95 | 253.14 | 232.78 |
| IV | 1425 | 50 | 60 | 10 | 40 | 140 | 119 | 94 | 58 | 27 | 31.33 | 25.72 | 19.98 | 13.20 | 5.92 | 10360.09 | 273.25 | 310.26 | 259.94 | 237.31 |
| v | 1410 | 20 | 60 | 20 | 30 | 99 | 77 | 58 | 35 | 15 | 22.69 | 17.30 | 12.33 | 7.87 | 3.34 | 10368.73 | 281.67 | 317.91 | 265.27 | 239.89 |
| vI | 1420 | 10 | 40 | 70 | 15 | 45 | 32 | 23 | 15 | 6 | 10.16 | 7.35 | 5.03 | 3.14 | 1.36 | 10381.26 | 291.62 | 325.21 | 270.00 | 242.87 |

Table 3. Humidities and Densities of air in suction pipe.

| $\underset{\text { resistance }}{\substack{\text { Fan }}}$ | Relative humidity $p_{s}$ by (5) |  |  |  |  | Air constant $\mathrm{R}_{\mathrm{s}}$ by (7) |  |  |  |  | Air density $\gamma_{s}$ by (9) in $\mathrm{kg} / \mathrm{m}^{3}$ |  |  |  |  | $\sqrt{\frac{2 \mathrm{~g}}{\gamma_{s}}}$ in (10) |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 |
| $\bigcirc$ |  |  |  |  |  |  |  |  |  |  | 1.265 | 1.246 | 1.249 | 1.243 | 1.219 |  |  |  |  |  |
| I |  |  |  |  |  |  |  |  |  |  | 1.265 | 1246 | 1.249 | 1243 | 1.219 | 3.936 | 3.966 | 3.961 | 3.971 | 4.010 |
| 11. |  |  |  |  |  |  |  |  |  |  | 1.265 | 1.246 | 1.249 | 1.243 | 1.219 | 3.936 | 3.966 | 3.961 | 3.971 | 4.010 |
| III | 0.72 | 0.80 | 0.61 | 0.61 | 0.57 | 29.34 | 29.36 | 29.34 | 29.34 | 29.23 | 1.265 | 1.247 | 1.250 | 1.244 | 1.220 | 3.936 | 3.965 | 3.960 | 3.969 | 4.008 |
| IV |  |  |  |  |  |  |  |  |  |  | 1.266 | 1.247 | 1.251 | 1.244 | 1.221 | 3.935 | 3.965 | 3.958 | 3.969 | 4.006 |
| $v$ |  |  |  |  |  |  |  |  |  |  | 1.267 | 1.248 | 1.252 | 1.245 | 1.221 | 3.934 | 3.963 | 3957 | 3.968 | 4.006 |
| vi |  |  |  |  |  |  |  |  |  |  | 1.268 | 1.250 | - 1.253 | 1.246 | 1.222 | 3.931 | 3.961 | 3.955 | 3.966 | 4.005 |

Table 4－1．Air velocities and Volumes measured in Suction pipe．（Fan 1．）

| Measuring prsition in fig． 9 | Fan speed in rev／mn |  |  |  |  |  | Reading in $\cdot \mathrm{mm}$ of alcohol（ 0.835 density），by manometer inclination $=$ |  |  |  |  |  | $\mathrm{h}_{\boldsymbol{v}}$ in（10）reduced to $1450 \mathrm{rev} / \mathrm{mn}$ in mm Aq or $\mathrm{kg} / \mathrm{ni}^{2}$ |  |  |  |  |  | Air velocity，by（10）in m， s ． |  |  |  |  |  | Calculated results． |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  | $0^{\circ} 35^{\prime}$ | $1^{\circ} 10^{\prime}$ | $1^{\circ} 10^{\prime}$ | $5^{\circ} 45^{\prime}$ | $5^{\circ} 45^{\prime}$ | $5^{\circ} 45^{\prime}$ |  |  |  |  |  |  |  |  |  |  |  |  |  | 药 |  |  |
|  | 1 | II | III | IV | $v$ | VI | 1 | II | II！ | IV | v | VI | 1 | II | III | IV | v | VI | 1 | 11 | III | IV | v | vI | $\stackrel{\circ}{2}$ | $\begin{gathered} 9 \\ \stackrel{y}{6} \\ \hline \end{gathered}$ |  |  |
| $1^{\prime}$ | 1440 | 1465 | 1440 | 1480 | 1420 | 1410 | 10 | 16 | 35 | 12 | 16 | 21 | 0.085 | 0.262 | 0.593 | 0.958 | 1.391 | 1.850 | 1.15 | 2.02 | 3.03 | 3.85 | 4.64 | 5.35 |  | $\ddagger$ |  |  |
| $2^{\prime}$ | 1440 | 65 | 40 | 80 | 20 | 20 | 11 | 16 | 40 | 20 | 24 | 30 | 0.093 | 0.262 | 0.678 | 1.596 | 2.085 | 2.610 | 1.20 | 2.02 | 3.24 | 4.97 | 5.69 | 6.35 | $4$ | $\underset{\Xi}{E}$ |  |  |
| $3^{\prime}$ | 1440 | 60 | 10 | 80 | 20 | 50 | 11 | 23 | 52 | 22 | 27 | 35 | 0.093 | 0.378 | 0.917 | 1.756 | 2.344 | 2.923 | 1.20 | 2.42 | 3.76 | 5.22 | 6.02 | 6.72 | 5 | ${ }_{5}$ | 言 |  |
| $4^{\prime}$ | 1480 | 60 | 80 | 80 | 20 | 50 | 14 | 25 | 61 | 22 | 30 | 39 | 0.112 | 0.413 | 0.973 | 1.756 | 2.610 | 3.257 | 1.33 | 2.53 | 3.89 | 522 | 6.35 | 7.10 | $\Xi$ |  | ＂ |  |
| $5^{\circ}$ | 1480 | 40 | 80 | 70 | 50 | 50 | 14 | 26 | 64 | 23 | 31 | 42 | 0.112 | 0.441 | 1．022 | 1.869 | 2.589 | 3.507 | 1.32 | 2.62 | 3.88 | 5.38 | 6.33 | 7.36 | $\Xi$ |  | ＋${ }^{*}$ |  |
| $6^{\prime}$ | 1480 | 40 | 80 | 70 | 50 | 50 | 15 | 26 | 65 | 23 | 32 | 44 | 0.120 | 0.441 | 1.037 | 1.869 | 2.672 | ＇3．675 | 1.36 | 2.62 | 4.01 | 5.38 | 6.43 | 7.54 | $\stackrel{\circ}{3}$ | 8 | $\stackrel{\text {－}}{\substack{\text { ® }}}$ | $\square$ |
| 7 | 1480 | 40 | 80 | 70 | 50 | 30 | 15 | 26 | 66 | 24 | 32 | 42 | 0.120 | 0.441 | 1.054 | 1.950 | 2.672 | 3.608 | 1.36 | 2.62 | 4.04 | 5.49 | 6.43 | 7.47 |  |  |  |  |
| $8{ }^{\prime}$ | 1460 － | 40 | 80 | 70 | 50 | 30 | 15 | 26 | 66 | 23 | 32 | 42 | 0.124 | 0.441 | 1.054 | 1.869 | 2.672 | 3.608 | 1.38 | 2.62 | 4.04 | 5.38 | 6.43 | 7.47 | 앙 | $\bigcirc$ | － | $\pm$ |
| c | 1460 | 40 | 40 | 50 | 50 | 50 | 15 | 26 | 62 | 22 | 32 | 44 | 0.124 | 0.441 | 1.050 | 1.837 | 2.672 | 3.675 | 1.38 | 2.62 | 4.04 | 5.33 | 6.43 | 7.54 | － | 8 |  |  |
| 8 | 1460 | 35 | 40 | 50 | 50 | 50 | 15 | 26 | 62 | 22 | 32 | 44 | 0.124 | 0.444 | 1.050 | 1.837 | 2.672 | 3.675 | 1.38 | 2.62 | 4.04 | 5.33 | 6.43 | 7.54 |  |  | $\stackrel{\infty}{0}$ |  |
| 7 | 1440 | 35 | 40 | 40 | 50 | 50 | 14 | 26 | 61 | 21 | 32 | 44 | 0.119 | 0.444 | 1.032 | 1.778 | 2.672 | 3.675 | 1.36 | 2.62 | 3.99 | 5.24 | 6.43 | 7.54 | \％ | \％ |  | E |
| 6 | 1440 | 35 | 40 | 40 | 50 | 50 | 14 | 26 | 60 | 21 | 32 | 44 | 0.119 | 0.444 | 1.016 | 1.778 | 2.672 | 3.675 | 1.36 | 2.62 | 3.97 | 5.24 | 6.43 | 7.54 |  |  |  |  |
| 5 | 1440 | 35 | 40 | 40 | 60 | 50 | 14 | 26 | 60 | 21 | 31 | 44 | 0.119 | 0.444 | 1.016 | 1.778 | 2.556 | 3.675 | 1.36 | 2.62 | 3.97 | 5.24 | 6.29 | 7.54 | 菏 | 总 | 8 | 4 |
| 4 | 1430 | 40 | 65 | 40 | 60 | 50 | 12 | 25 | 58 | 20 | 30 | 42 | 0.103 | 0.425 | 0.949 | 1.693 | 2.470 | 3.507 | 1.26 | 2.57 | 3.84 | 5.12 | 6.22 | 7.36 |  |  |  |  |
| 3 | 1430 | 40 | 65 | 40 | 60 | 50 | 11 | 22 | 56 | 19 | 28 | 39 | 0.095 | 0.373 | 0.916 | 1.608 | 2.305 | 3.257 | 1.21 | 2.40 | 3.76 | 4.99 | 5.98 | 7.10 | 皆 | 会 | $\stackrel{\circ}{\circ}$ | 4 |
| 2 | 1410 | 40 | 70 | 60 | 50 | 50 | 10 | 16 | 48 | 18 | 24 | 35 | 0.088 | 0.271 | 0.781 | 1.484 | 2.004 | 2.923 | 1.17 | 2.05 | 3.48 | 4.80 |  |  |  |  |  |  |
| 1 | 1480 | 40 | 40 | 60 | 50 | 50 | 11 | 16 | 35 | 12 | 17 | 25 | 0.083 | 0.271 | 0.593 | 0.989 | 1.420 | 2.087 | 1.13 | 2.05 | 3.03 | 3.91 | 4.69 | 5.68 | － | 里 | － | $\leq$ |

Table 4－2．Air velocities and Volumes measured in suction pipe．（Fan 2．）

| Measuring position in fig． 9 | Fan speed in rev／mn |  |  |  |  |  | Reading in mm of alcohol（ 0.834 density），by manometer inclination $=$ |  |  |  |  |  | $h_{v}$ in（10）reduced to $1450 \mathrm{rev} / \mathrm{mn}$ in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$ |  |  |  |  |  | Air velocity，by（10）in m／s． |  |  |  |  |  | Calculated results． |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  | $0^{\circ} 35^{\prime}$ | $1^{\circ} 10^{\prime}$ | $1^{\prime} 10^{\prime}$ | $1^{\circ} 10^{\prime}$ | $5^{\circ} 45^{\prime}$ | $5^{\circ} 45^{\prime}$ |  |  |  |  |  |  |  |  |  |  |  |  |  | 을 |  |  |
|  | 1 | II | III | IV | $v$ | 1 | 1 | II | III | 1v | v | vi | 1 | II | II | IV | $v$ | v1 | 1 | II | III | IV | v | VI | $\stackrel{\text { a }}{ }$ | $\begin{aligned} & \text { \% } \\ & < \end{aligned}$ |  |  |
| $1^{\prime}$ | 1450 | 1475 | 1475 | 1465 | 1470 | 1420 | 8 | 15 | 30 | 46 | 13 | 15 | 0.067 | 0.241 | 0.484 | 0.752 | 1.056 | 1.307 | 1.02 | 1.95 | 2.76 | 3.44 | 4.07 | 4.53 |  | $\square$ |  |  |
| $2^{\prime}$ | 1450 | 75 | 75 | 60 | 70 | 20 | 8 | 19 | 44 | 71 | 18 | 21 | 0.067 | 0.306 | 0.698 | 1.167 | 1，460 | 1825 | 1.02 | 2.19 | 3.32 | 4.28 | 4.79 | 5.34 | $4$ | $\underset{\Xi}{\Xi}$ |  |  |
| $3^{\prime}$ | 1450 | 75 | 65 | 60 | 70 | 30 | 9 | 23 | 52 | 79 | 22 | 25 | 0.075 | 0.371 | 0.850 | 1.298 | 1.785 | 2.140 | 1.08 | 2.42 | 3.66 | 4.52 | 5.29 | 5.78 | 5 | $\stackrel{3}{\mathrm{~g}}$ | 宫 |  |
| $4^{\prime}$ | 1450 | 75 | 50 | 70 | 60 | 30 | 12 | 26 | 54 | 85 | 23 | 29 | 0.100 | 0.418 | 0.902 | 1.380 | 1.892 | 2.490 | 1.25 | 2.57 | 3.77 | 4.66 | 5.45 | 6.25 | $\Xi$ |  | 1\％ |  |
| $5^{\prime}$ | 1450 | 90 | 55 | 70 | 60 | 50 | 12 | 27 | 57 | 92 | 24 | 30 | 0.100 | 0.427 | 0.945 | 1.493 | 1.970 | 2.502 | 1.25 | 2.59 | 3.86 | 4.85 | 5.57 | 6.26 | $\Xi$ |  | ＋${ }^{1}$ |  |
| $6^{\prime}$ | 1450 | 60 | 55 | 70 | 60 | 50 | 12 | 26 | 57 | 93 | 24 | 30 | 0.100 | 0.428 | 0.945 | 1.510 | 1.970 | 2.502 | 1.25 | 2.60 | 3.86 | 4.88 | 5.57 | 6.26 |  |  | $\stackrel{\square}{\square}$ |  |
| 7 | 1460 | 60 | 70 | 70 | 60 | 50 | 12.5 | 26 | 58 | 94 | 24 | 31 | 0.140 | 0.428 | 0.943 | 1.525 | 1.970 | 2.585 | 1.29 | 2.60 | 3.85 | 4.90 | 5.57 | 6.37 | \％ | \％ | $\underset{\sim}{\infty}$ | － |
| $8^{\prime}$ | 1460 | 55 | 70 | 70 | 45 | 50 | 12.5 | 26 | 58 | 94 | 24 | 31 | 0.140 | 0.430 | 0.943 | 1.525 | 2.014 | 2.585 | 1.29 | 2.61 | 3.85 | 4.90 | 5.62 | 6.37 |  |  |  |  |
| C | 1450 | 40 | 70 | 65 | 45 | 50 | 12.5 | 26 | 58 | 94 | 24 | 31 | 0.142 | 0．440 | 0.943 | 1.537 | 2.014 | 2.585 | 1.30 | 2.63 | 3.85 | 4.92 | 5.62 | 6.37 | \％ | \％ | $\pm$ | ＝ |
| 8 | 1450 | 40 | 70 | 60 | 45 | 50 | 12.5 | 26 | 58 | 92 | 24 | 31 | 0.142 | 0.440 | 0.943 | 1.513 | 2.014 | 2.585 | 1.30 | 2.63 | 3.85 | 4.88 | 5.62 | 6.37 |  |  |  |  |
| 7 | 1455 | 40 | 65 | 60 | 45 | 50 | 12.5 | 26 | 57 | 92 | 24 | 30 | 0.141 | 0.440 | 0.933 | 1.513 | 2.014 | 2.502 | 1.29 | 2.63 | 3.83 | 4.88 | 5.62 | 6.25 | \％ | ® | 8 | $\xi$ |
| 6 | 1455 | 50 | 65 | 60 | 45 | 50 | 12 | 26 | 56 | 92 | 24 | 30 | 0.099 | 0.434 | 0.915 | 1.513 | 2.014 | 2.502 | 1.25 | 2.61 | 3.79 | 4.88 | 5.62 | 6.25 |  |  |  |  |
| 5 | 1455 | 50 | 65 | 40 | 45 | 50 | 12 | 26 | 56 | 87 | 23 | 30 | 0.099 | 0.434 | 0.915 | 1.470 | 1.930 | 2.502 | 1.25 | 2.61 | 3.79 | 4.82 | 5.51 | 6.25 | \＆ | \％ | \％ | 8 |
| 4 | 1460 | 50 | 65 | 40 | 45 | 50 | 11 | 25 | 53 | 86 | 23 | 29 | 0.090 | 0.417 | 0.867 | 1.455 | 1.930 | 2.420 | 1.19 | 2.56 | 3.69 | 4.78 | 5.51 | 6.15 |  |  |  |  |
| 3 | 1460 | 60 | 65 | 35 | 60 | 50 | 9 | 22 | 50 | 75 | 21 | 26 | 0.074 | 0.363 | 0.817 | 1.278 | 1.724 | 2.170 | 1.08 | 2.39 | 3.58 | 4.48 | 5.22 | 5.83 | 薥． | － | 皆 | 4 |
| 2 | 1460 | 60 | 65 | 25 | 60 | 40 | 8 | 18 | 43 | 69 | 18 | 22 | 0.066 | 0.296 | 0.704 | 1.190 | 1.480 | 1.860 | 1.02 | 2.16 | 3.33 | 4.33 | 4.83 | 5.40 |  |  |  |  |
| 1 | 1460 | 60 | 70 | 60 | 55 | 40 | 8 | 15 | 31 | 47 | 13 | 16 | 0.066 | 0.246 | 0.503 | 0.773 | 1.077 | 1.352 | 1.02 | 1.97 | 2.82 | 3.49 | 4.11 | 4.61 | － | 感 | \％ | $\leq$ |

Table 4－3．Air velocities and Volumes measured in suction pipe．（Fan 3．）

| Measuring position in fig． 9 | Fan speed in rev／mn |  |  |  |  |  | Reading in mm of alcohol（ 0.833 density），by manometer inclination $=$ |  |  |  |  |  | $\mathrm{h}_{v}$ in（ t 0 ）reduced to $\mathrm{I} 450 \mathrm{rev} / \mathrm{mn}$ in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$ |  |  |  |  |  | Air velocity，by（ x ）in m／s． |  |  |  |  |  | Calculated resilts． |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  | $0^{\circ} 35^{\prime}$ | $0^{\prime} 35^{\prime}$ | $1^{\circ} 10^{\prime}$ | $1^{\circ} 10^{\prime}$ | $1^{\circ} 10^{\prime}$ | $5^{c} 45^{\prime}$ |  |  |  |  |  |  |  |  |  |  |  |  |  | 台 | $\left\|\begin{array}{c} x \\ 3 \end{array}\right\|$ |  |
|  |  |  |  |  |  |  | 1 | II | III | Iv | v | vI |  |  |  |  | $\checkmark$ | I | 1 | н | II | 1 | $\checkmark$ | I | $\stackrel{2}{2}$ | $\overline{0}$ | ¢ |  |
| $1^{\prime}$ | 1455 | 1460 | 1460 | 1480 | 1450 | 1420 | 6 | 26 | 23 | 39 | 44 | 10 | 0.050 | 0.214 | 0.378 | 0.621 | 0.733 | 0.845 | 0.88 | 1.83 | 2.44 | 3.12 | 3.39 | 3.64 | 亮 | $5$ | －${ }^{\circ}$ |  |
| $2^{\prime}$ | 1455 | 40 | 60 | 65 | 50 | 40 | 6.5 | 33 | 33 | 55 | 65 | 15 | 0.054 | 0.279 | 0.542 | 0.898 | 1.083 | 1.267 | 0.92 | 2.12 | 2.91 | 3.75 | 4.12 | 4.45 | 4 | $\Xi$ | 边 |  |
| $3^{\prime}$ | 1480 | 40 | 55 | 65 | 50 | 40 | 7.5 | 40 | 40 | 61 | 77 | 17 | 0.060 | 0.338 | 0.662 | 0.996 | ． 1.283 | 1.436 | 0.97 | 2.31 | 3.22 | 3.95 | 4.49 | 4.74 | 5 | \％ | －宫 $\frac{5}{4}$ |  |
| $4^{\prime}$ | 1480 | 35 | 70 | 65 | 60 | 40 | 10 | 44 | 42 | 65 | 81 | 18 | 0.080 | 0.374 | 0.680 | 1.061 | 1.331 | 1.520 | 1.12 | 2.43 | 3.27 | 4.08 | 4.56 | 4.87 | $\Xi$ |  | ＂ |  |
| $5^{\prime}$ | 1480 | 40 | 70 | 60 | 60 | 40 | 10 | 46 | 44 | 68 | 86 | 19 | 0.080 | 0.389 | 0.713 | 1.118 | 1.414 | 1.605 | 1.12 | 2.47 | 3.25 | 4.18 | 4.71 | 5.02 | 5 |  | ＋${ }^{\prime \prime}$ |  |
| $6^{\prime}$ | 1480 | 40 | 70 | 60 | 60 | 40 | 10 | 46 | 44 | 69 | 87 | 20 | 0.080 | 0.389 | 0.713 | 1.130 | 1.430 | 1.689 | 1.12 | 2.47 | 3.25 | 4.21 | 4.74 | 5，14 | 8 | 8 | $\stackrel{\circ}{\circ}$ | H |
| 7 | 1460 | 40 | 60 | 60 | 60 | 40 | 10 | 46 | 44 | 69 | 87 | 20 | 0.082 | 0.389 | 0.723 | 1.130 | 1.430 | 1.689 | 1.14 | 2.47 | 3.37 | 4.21 | 4.74 | 5.14 | \％ | ¢ |  | － |
| $8^{\prime}$ | 1460 | 40 | 60 | 50 | 60 | 40 | 10 | 46 | 44 | 67 | 86 | 20 | 0.082 | 0.389 | 0.723 | 1.162 | 1.414 | 1.689 | 1.14 | 2.47 | 3.37 | 4.27 | 4.71 | 5.14 |  |  | No | $E$ |
| C | 1480 | 40 | ${ }_{0} 0$ | 70 | 60 | 40 | 10 | 46 | 43 | 69 | 86 | 20 | 0.080 | 0.389 | 0.726 | 1.119 | 1.414 | 1.689 | 1.12 | 2.47 | 3.38 | 4.19 | 4.71 | 5.14 | \％ | 胣• |  | $=$ |
| 8 | 1480 | 40 | 40 | 75 | 60 | 40 | 10 | 46 | 43 | 69 | 86 | 20 | 0.080 | 0.389 | 0.726 | 1.111 | 1.414 | 1.689 | 1.12 | 2.47 | 3.38 | 4.17 | 4.71 | 5.14 |  |  |  |  |
| 7 | 1480 | 40 | 30 | 75 | 60 | 40 | 10 | 46 | 42 | 69 | 86 | 20 | 0.080 | 0.389 | 0.720 | 1.111 | 1.414 | 1.689 | 1.12 | 2.47 | 3.36 | 4.17 | 4.71 | 5.14 | 㲾 | 箅 | $\dot{\circ}$ | 日 |
| 6 | 1480 | 40 | 25 | 90 | 60 | 40 | 10 | 46 | 41 | 69 | 86 | 20 | 0.080 | 0.389 | 0.707 | 1.088 | 1.414 | 1.689 | 1.12 | 2.47 | 3.33 | 4.17 | 4.71 | 5.14 |  |  |  |  |
| 5 | 1470 | 40 | 25 | 60 | 60 | 30 | 10 | 46 | 41 | 66 | 86 | － 20 | 0.081 | 0.389 | 0.707 | 1.085 | 1.414 | 1.712 | 1.13 | 2.47 | 3.33 | 4.16 | 4.71 | 5.17 | 寫 | 墻 | \％ | Z |
| 4 | 1470 | 25 | 30 | 60 | 60 | 30 | 10 | 43 | 41 | 63 | 81 | 20 | 0.081 | 0.371 | 0.702 | 1.035 | 1.331 | 1.712 | 1.13 | 2.41 | 3.32 | 4.03 | 4.56 | 5.17 |  |  |  |  |
| 3 | 1470 | 25 | 30 | 60 | 60 | 40 | 7.5 | 42 | 38 | 58 | 77 | 18 | 0.061 | 0.362 | 0.651 | 0.953 | 1.265 | 1.520 | 0.98 | 2.39 | 3.19 | 3.87 | 4.45 | 4.87 | 葛 | 蠋 | 聰 | 4 |
| 2 | 1460 | 25 | 30 | 45 | 60 | 40 | 6.5 | 30 | 31 | 50 | 65 | 15 | 0.053 | 0.259 | 0，531 | 0.838 | 1.068 | 1.267 | 0.92 | 2.05 | 2.88 | 3.63 | 4.09 | 4.45 |  |  |  |  |
| 1 | 1460 | 25 | 30 | 45 | 65 | 40 | 6 | 27 | 23 | 37 | 46 | 10 | 0.049 | 0.233 | 0.394 | 0.620 | 0.760 | 0.845 | 0.88 | 1.91 | 2.49 | 3.12 | 3.45 | 3.64 | 产 | 象 | ．${ }_{-\infty}$ | $\leq$ |

Table 4－4．Air velocities and Volumes measured in Suction pipe．（Fan 4．）

| Measuring position in fig． 9 | Fan speed in rev／mn |  |  |  |  |  | Reading in mm of alcohol（o．833 density），by manometer inclination＝ |  |  |  |  |  | $\mathrm{h}_{v}$ in（10）reduced to $1450 \mathrm{rev} / \mathrm{mn}$ in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$ |  |  |  |  |  | Air velocity，by（10）in m／s． |  |  |  |  |  | Calculated results． |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  | $0^{\circ} 35^{\prime}$ | $1^{\circ} 10^{\prime}$ | $1^{\circ} 10^{\prime}$ | $1^{\circ} 10^{\prime}$ | $1^{\circ} 10^{\prime}$ | $1^{\circ} 10^{\prime}$ |  |  |  |  |  |  |  |  |  |  |  |  |  | $\frac{4}{4}$ |  |  |
|  |  |  |  |  |  | v | I | 11 | III | IV | v | vi | 1 | п1 | If | Ir | $v$ | V | 1 | 1 | III | IV | $v$ | V1 | $\stackrel{\circ}{2}$ | $$ | 年 |  |
| $1^{\prime}$ | 1465 | 1445 | 1460 | 1470 | 1490 | 1475 | 6 | 11 | 17 | 24 | 29 | 31 | 0.049 | 0.184 | 0.280 | 0.389 | 0.457 | 0.498 | 0.89 | 1.70 | 2.10 | 2.48 | 2.68 | 2.80 | $\frac{0}{1}$ | $\square$ |  |  |
| ${ }^{\prime}$ | 1440 | 45 | 60 | 70 | 80 | 75 | 6 | 14 | 25 | 36 | 41 | 42 | 0.051 | 0.235 | 0.410 | 0.584 | 0.653 | 0.676 | 0.89 | 1.92 | 2.54 | 3.03 | 3.21 | 3.26 | 4 | $\underset{\Xi}{\Xi}$ |  |  |
| $3^{\prime}$ | 1450 | 55 | 60 | 70 | 80 | 70 | 7 | 17 | 29 | 39 | 46 | 48 | 0.058 | 0.282 | 0.478 | 0.633 | 0.732 | 0.779 | 0.96 | 2.11 | 2.74 | 3.16 | 3.39 | 3.50 | ¢ | 怱 | 产 $\frac{3}{4}$ |  |
| $4^{\prime}$ | 1475 | 55 | 40 | 55 | 80 | 70 | 9 | 18 | 31 | 42 | 50 | 54 | 0.072 | 0.298 | 0.524 | 0.695 | 0.796 | 0.876 | 1.07 | 2.17 | 2.88 | 3.31 | 3.54 | 3.71 | $\Xi$ |  | ＂ |  |
| $5^{\prime}$ | 1475 | 55 | 40 | 40 | 80 | 80 | 10 | 19 | 32 | 43 | 53 | 58 | 0.080 | 0.315 | 0.541 | 0.726 | 0.844 | 0.924 | 1.13 | 2.23 | 2.92 | 3.39 | 3.64 | 3.77 | 3 |  | ＋${ }^{*}$ |  |
| $6^{\prime}$ | 1475 | 55 | 40 | 40 | 80 | 90 | 10 | 19 | 32 | 43 | 53 | 60 | 0.080 | 0.315 | 0.541 | 0.726 | 0.844 | 0.947 | 1.13 | 2.23 | 2.92 | 3.39 | 3.64 | 3.85 |  |  | $\bigcirc$ |  |
| 7 | 1460 | 55 | 40 | 30 | 80 | 90 | 10 | 19 | 32 | 44 | 55 | 60 | 0.082 | 0.315 | 0.541 | 0.754 | 0.876 | 0.947 | 1.14 | 2.23 | 2.92 | 3.45 | 3.71 | 3.85 | 8 | $\stackrel{\text { ¢ }}{\text { G }}$ | 8 | $\rightarrow$ |
| $8^{\prime}$ | 1460 | 55 | 40 | 30 | 80 | 90 | 10 | 19 | 32 | 44 | 55 | 60 | 0.082 | 0.315 | 0.541 | 0.754 | 0.876 | 0.947 | 1.14 | 2.23 | 2.92 | 3.45 | 3.71 | 3.85 |  |  |  |  |
| C | 1460 | 70 | 40 | 30 | 70 | 80 | 10 | 19 | 32 | 44 | 53 | 60 | 0.082 | 0.309 | 0.541 | 0.754 | 0.859 | 0.956 | 1.14 | 2.20 | 2.92 | 3.45 | 3.68 | 3.88 | 尔 | \％ | \％ | $\square$ |
| 8 | 1460 | 70 | 40 | 40 | 70 | 80 | 10 | 19 | 32 | 44 | 53 | 60 | 0.082 | 0.309 | 0.541 | 0，743 | 0.859 | 0.956 | 1.14 | 2.20 | 2.92 | 3.42 | 3.68 | 3.88 |  |  | $\stackrel{\sim}{\circ}$ |  |
| 7 | 1450 | 70 | 40 | 40 | 70 | 80 | 10 | 19 | 32 | 44 | 53 | 60 | 0.083 | 0.309 | 0.541 | 0.743 | 0.859 | 0956 | 1.15 | 2.20 | 2.92 | 3.42 | 3.68 | 3.88 | 器 | $\stackrel{0}{\infty}$ | \％ | 日 |
| ${ }^{6}$ | 1450 | 80 | 40 | 40 | 70 | 80 | 10 | 19 | 31 | 44 | 53 | 60 | 0.083 | 0.303 | 0.524 | 0.743 | 0.859 | 0.956 | 1.15 | 2.18 | 2.88 | 3.42 | 3.68 | 3.88 |  |  |  |  |
| 5 | 1450 | 80 | 40 | 40 | 70 | 80 | 10 | 19 | 31 | 43 | 53 | 60 | 0.083 | 0.303 | 0.524 | 0.726 | 0.859 | 0.956 | 1.15 | 2.18 | 2.88 | 3.38 | 3.68 | 3.88 | \％ | 咢 | N | 8 |
| 4 | 1450 | 80 | 50 | 50 | 70 | 90 | 10 | 18 | 30 | 42 | 50 | 60 | 0.083 | 0.287 | 0.500 | 0.700 | 0.811 | 0.947 | 1.15 | 2.13 | 2.81 | 3.32 | 3.57 | 3.85 |  |  |  |  |
| 3 | 1450 | 55 | 50 | 50 | 70 | 90 | 8 | 16 | 27 | 38 | 47 | 54 | 0.067 | 0.265 | 0.450 | 0.633 | 0.762 | 0.852 | 1.02 | 2.05 | 2.66 | 3.16 | 3.46 | 3.66 | \％ | 过 | $\stackrel{\text { ¢ }}{ \pm}$ | 4 |
| 2 | 1440 | 55 | 50 | 50 | 70 | 90 | 7 | 14 | 22 | 37 | 38 | 43 | 0.059 | 0.231 | 0.360 | 0.617 | 0.616 | 0.678 | 0.97 | 1.91 | 2.40 | 3.12 | 3.09 | 3.26 |  |  |  |  |
| 1 | 1440 | 55 | 50 | 50 | 70 | 80 | 6 | 12 | 18 | 25 | 30 | 33 | 0.051 | 0.199 | 0.300 | 0.416 | 0.487 | 0.526 | 0.90 | 1.77 | 2.18 | 2.56 | 2.77 | 2.88 | \％ | 旡 | \％ | $\leq$ |

Table 4－5．Air velocities and Volumes measured in suction pipe．（Fan 5．）

| Measuring position in fig． 9 | Fan speed in rev／mn |  |  |  |  |  | Reading in mm of alcohol（ 0.836 density），by manometer inclination $=$ |  |  |  |  |  | $\mathrm{h}_{\boldsymbol{v}}$ in（10）reduced to $1450 \mathrm{rev} / \mathrm{mn}$ in mm Aq or $\mathrm{kg} \mathrm{m}{ }^{2}$ |  |  |  |  |  | Air velocity，by（10）in m／s． |  |  |  |  |  | Calculated results． |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | I | II | II | IV | v | vI | $0^{\circ} 35^{\prime}$ | $0^{\circ} 35^{\prime}$ | $035^{\prime}$ | $1^{\circ} 10^{\prime}$ | $1^{0} 10^{\prime}$ | $1^{\circ} 10^{\prime}$ | 1 | II | III | IV | v | vi | ，I | II | III | IV | v | vI | $\frac{\tilde{g}_{0}^{2}}{0}$ | $\begin{aligned} & 4 \\ & \frac{4}{3} \end{aligned}$ |  |  |
|  |  | I |  |  |  |  | I | II | III | IV | v | vi |  |  |  |  |  |  |  |  |  |  |  |  | $\begin{aligned} & \tilde{i} \\ & = \end{aligned}$ | ${ }_{4}^{2}$ |  |  |
| $1^{\prime}$ | 1460 | 1460 | 1420 | 1460 | 1450 | 1420 | 5 | 13 | 14 | 11 | 11 | 11 | 0.041 | 0.107 | 0.122 | 0.182 | 0.184 | 0.192 | 0.81 | 1.32 | 1.40 | 1.71 | 1.72 | 1.75 | 曾 | $\stackrel{5}{5}$ |  |  |
| ${ }^{\prime \prime}$ | 1460 | 90 | 20 | ${ }^{60}$ | 70 | 40 | 5 | 18 | 21 | 14 | 16 | 14 | 0.041 | 0.143 | 0.183 | 0.231 | 0.260 | 0.237 | 0.81 | 1.51 | 1.71 | 1.93 | 2.04 | 1.95 | $4$ | $\underset{\Xi}{\Xi}$ | 边 |  |
| $3^{\prime}$ | 1460 | 70 | 30 | 60 | 70 | 60 | 5.5 | 19 | 25 | 17 | － 18 | 17 | 0.045 | 0.155 | 0.215 | 0.280 | 0.292 | 0.280 | 0.85 | 1.58 | 1.86 | 2.12 | 2.17 | 2.12 | 5 | 邑 | － |  |
| $4 \prime$ | 1460 | 70 | 30 | 25 | 70 | 60 | 7.5 | 20 | 29 | 17 | 20 | 20 | 0.062 | 0.163 | 0.246 | 0.294 | 0.325 | 0.330 | 1.00 | 1.62 | 2.00 | 2.17 | 2.28 | 2.30 | $\Xi$ |  | ＂ |  |
| $5^{\prime}$ | 1460 | 70 | 35 | 25 | 70 | 60 | 8 | 21 | 29 | 18 | 20 | 21 | 0.066 | 0.171 | 0.247 | 0.311 | 0.325 | 0.346 | 1.03 | 1.67 | 1.99 | 2.24 | 2.28 | 2.35 | $\xi$ |  | ＋${ }^{\circ} \mathrm{E}$ |  |
| $6^{\prime}$ | 1460 | 70 | 35 | 50 | 60 | 60 | 8 | 21 | 29 | 19 | 20 | 21 | 0.066 | 0.171 | 0.247 | 0.317 | 0.330 | 0.346 | 1.03 | 1.67 | 1.99 | 2.26 | 2.30 | 2.35 | 8 | 8 | 8 | ${ }^{-}$ |
| $7{ }^{\prime}$ | 1470 | 40 | 35 | 50 | 60 | 40 | 8 | 21 | 29 | 19 | 20 | 21 | 0.065 | 0.178 | 0.247 | 0.317 | 0.330 | 0.356 | 1.02 | 1.69 | 1.99 | 2.26 | 2.30 | 2.39 | \％ | － |  | － |
| $8^{\prime}$ | 1470 | 60. | 40 | 50 | 60 | 40 | 8 | 22 | 30 | 19 | 20 | 21 | 0.065 | 0.182 | 0.255 | 0.317 | 0.330 | 0.356 | 1.02 | 1.71 | 2.02 | 2.26 | 2.30 | 2.39 |  | $\bigcirc$ | $\div$ | $\pm$ |
| C | 1470 | 60 | 60 | 30 | 60 | 40 | 8 | 22 | 31 | 18 | 20 | 21 | 0.065 | 0.182 | 0256 | 0.309 | 0.330 | 0.356 | 1.02 | 1.71 | 2.02 | 2.23 | 2.30 | 2.39 | 宛 | 晏 | \％ | $\pm$ |
| 8 | 1470 | 60 | 60 | 30 | 60 | 40 | 8 | 22 | 31 | 18 | 20 | 21 | 0.065 | 0.182 | 0.256 | 0.309 | 0.330 | 0.356 | 1.02 | 1.71 | 2.02 | 2.23 | 2.30 | 2.39 | $\bigcirc$ |  | $\checkmark$ |  |
| 7 | 1455 | 60 | 60 | 30 | 60 | 30 | 8 | 21 | 31 | 18 | 20 | 21 | 0.066 | 0.173 | 0.256 | 0.309 | 0.330 | 0.361 | 1.03 | 1.67 | 2.02 | 2.23 | 2.30 | 2.40 | ¢ | 항 | $\infty$ | \＃ |
| 6 | 1455 | 60 | 60 | 30 | 60 | 30 | 7.5 | 21 | 31 | 18 | 20 | 21 | 0.062 | 0.173 | 0.256 | 0.309 | 0.330 | 0.361 | 1.00 | 1.67 | 2.02 | 2.23 | 230 | 2.40 |  |  |  |  |
| 5 | 1455 | 60 | 45 | 20 | 60 | 30 | 7.5 | 21 | 30 | 18 | 20 | 21 | 0.062 | 0.173 | 0.253 | 0.313 | 0.330 | 0.361 | 1.00 | 1.67 | 2.01 | 2.24 | 2.30 | 2.40 | $\stackrel{3}{6}$ | \％ | ： | $\%$ |
| 4 | 1455 | 50 | 45 | 50 | 55 | 55 | 7 | 20 | 29 | 17 | 19 | 20 | 0.058 | 0.167 | 0.244 | 0.284 | 0.315 | 0.332 | 0.97 | 1.64 | 1.98 | 2.14 | 2.25 | 2.31 |  |  |  |  |
| 3 | 1440 | 50 | 45 | 40 | 55 | 55 | 6 | 19 | 25 | 16 | 18 | 19 | 0.051 | 0.159 | 0.210 | 0.271 | 0.298 | 0.315 | 0.90 | 1.60 | 1.84 | 2.08 | 2.19 | 2.25 | 呙 | 岂 | $\stackrel{\sim}{\infty}$ | $\checkmark$ |
| 2 | 1440 | 50 | 65 | 40 | 70 | 40 | 5.5 | 17 | 21 | 13 | 15 | 15 | 0.047 | 0.142 | 0.172 | 0.220 | 0.244 | 0.255 | 0.87 | 1.51 | 1.66 | 1.88 | 1.98 | 2.02 |  |  |  |  |
| 1 | 1440 | 50 | 65 | 40 | 70 | 40 | 5 | 13 | 17 | 11 | 12 | 12 | 0.042 | 0.109 | 0.139 | 0.187 | 0.195 | 0.203 | 0.83 | 1.32 | 1.50 | 1.73 | 1.77 | 1.80 | － | － | 皆 | $\leq$ |

Table 5. Theoretical isothermal powers.

| Fan resistance. | Suction velocity pressure as in example 12. im $m m A q$ or $\mathrm{kg} / \mathrm{m}^{2}$. |  |  |  |  | Absolute suction pressure $\mathrm{pr}_{1}$ in ( I 2 ) in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$. |  |  |  |  | Delivery velocity pressure as in example 12. in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$. Delivery area in $\mathrm{m}^{2}$. $=$ |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | - 4 | 5 | 0.023 | 0.019 | 0.016 | 0.012 | 0.0085 |
|  |  |  |  |  |  |  |  |  |  |  | 1 | 2 | 3 | 4 | 5 |
| I | 0.11 | 0.09 | 0.07 | 0.07 | 0.06 | 10355.01 | 10266.00 | 10300.20 | 10245.71 | 10220.57 | 1.31 | 1.52 | 1.81 | 3.05 | 4.94 |
| II | 0.38 | 0.38 | 0.34 | 0.28 | 0.16 | 10356.36 | 264.45 | 298.00 | 248.32 | 226.39 | 4.58 | 6.50 | 8.90 | 11.80 | 14.05 |
| III | 0.91 | 0.66 | 0.64 | 0.48 | 0.22 | 10356.24 | 266.78 | 303.59 | 253.62 | 233.00 | 10.84 | 14.24 | 16.50 | 20.15 | 19.35 |
| IV | 1.65 | 1.32 | 1.00 | 0.66 | 0.28 | 10361.74 | 274.57 | 311.26 | 260.60 | 237.59 | 19.63 | 22.80 | 25.60 | 28.10 | 24.43 |
| v | 2.34 | 1.82 | 1.25 | 0.75 | 0.30 | 10371.07 | 283.49 | 319.16 | 266.02 | 240.19 | 27.98 | 31.50 | 32.40 | 32.30 | 25.98 |
| VI | 3.20 | 2.22 | 1.46 | 0.85 | 0.31 | 10384.46 | 293.84 | 326.67 | 270.85 | 243.18 | 38.17 | 38.60 | 37.70 | 36.20 | 27.40 |

Table 5. Continued.

| Fan | Absolute delivery pressure $\mathrm{p}_{2}$ in ( $\mathbf{1 2}$ ) in $\mathrm{mm} \mathrm{Aq}_{7}$ or $\mathrm{kg} / \mathrm{m}^{2}$. |  |  |  |  | Total isothermal power E by (12) in P.S. |  |  |  |  | Effective isothermal power $E^{\prime}$ as in example 12. in P.S. |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 |
| I | 10392.73 | 10300.49 | 10332.05 | 10276.19 | 10248.17 | 0.052 | 0.043 | 0.036 | 0.034 | 0.028 | 0.050 | 0.041 | 0.034 | 0.031 | 0.023 |
| II | 10396.00 | 305.47 | 339.14 | 284.94 | 257.28 | 0.102 | 0.105 | 0.101 | 0.082 | 0.053 | 0.091 | 0.089 | 0.079 | 0.055 | 0.029 |
| III | 10402.25 | 313.21 | 346.74 | 293.29 | 262.58 | 0.183 | 0.178 | 0.146 | 0.115 | 0.059 | 0.142 | 0.123 | 0.089 | 0.056 | 0.020 |
| IV | 10411.05 | 321.25 | 356.84 | 301.24 | 267.66 | 0.265 | 0.227 | 0.191 | 0.139 | 0.068 | 0.156 | 0.119 | 0.079 | 0.043 | 0.012 |
| V | 10419.40 | 330.47 | 362.64 | 305.44 | 269.21 | 0.310 | 0.266 | 0.204 | 0.144 | 0.067 | 0.131 | 0.088 | 0.051 | 0.026 | 0.006 |
| VI | 10429.59 | 337.57 | 367.94 | 309.34 | 270.63 | 0.338 | 0.274 | 0.210 | 0.158 | 0.065 | 0.052 | 0.032 | 0.013 | 0.009 | - |

Table 6. Pressure differences between suction and delivery.

| Fan resistance | Total pressure difference $P$ as in example 14. in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$. |  |  |  |  | Effective pressure difference $P$ ' as in example 14. in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$. |  |  |  |  | Theoretical head $\mathrm{H}_{\mathrm{w}}$ as in example 15 . in mm Aq or $\mathrm{kg} / \mathrm{m}^{2}$. |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 |
| 0 | 38.68 | 34.36 | 31.26 | 27.60 | 22.00 | 38.68 | 34.36 | 31.26 | 27.60 | 22.00 | 47.04 | 46.37 | 46.45 | 46.18 | 45.27 |
| I | 37.72 | 34.49 | 31.84 | 30.45 | 27.56 | 36.41 | 32.97 | 30.04 | 27.43 | 22.62 | 50.56 | 50.18 | 50.75 | 51.85 | 53.15 |
| II | 39.64 | 41.02 | 41.14 | 36.62 | 30.89 | 35.06 | 34.52 | 32.24 | 24.82 | 16.84 | 53.64 | 54.23 | 55.92 | 57.35 | 58.60 |
| III | 46.02 | 46.43 | 43.15 | 39.67 | 29.58 | 35.18 | 31.86 | 26.65 | 19.52 | 10.23 | 57.23 | 57.97 | 59.35 | 60.87 | 60.85 |
| IV | 49.31 | 47.20 | 44.58 | 40.64 | 30.07 | 29.68 | 24.40 | 18.98 | 12.54 | 5.64 | 60.74 | 61.14 | 62.55 | 63.63 | 62.80 |
| v | 48.33 | 46.98 | 43.48 | 39.42 | 29.02 | 20.35 | 15.48 | 11.08 | 7.12 | 3.04 | 63.40 | 63.64 | 64.51 | 64.94 | 63.35 |
| VI | 45.13 | 43.73 | 41.27 | 38.49 | 28.45 | 6.96 | 5.13 | 3.57 | 2.29 | 1.05 | 66.15 | 65.30 | 65.96 | 65.96 | 63.80 |

Table 7. Fan input powers.

| Fan resistance | Moment $\mathrm{M}_{\mathrm{e}}$ in (15) in mkg, by leverage $=1 \mathrm{~m}$. |  |  |  |  | Fan speed in rev/mn. |  |  |  |  | Power $\mathrm{E}_{\mathrm{f}}$ reduced to $1450 \mathrm{rev} / \mathrm{mn}$ as in example 16. in P.S. |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 |
| $\bigcirc$ | 0.023 | 0.01 | - | - | - | 1460 | 1440 | - | - | - | 0.047 | 0.021 | - | - | - |
| I | 0.038 | 0.033 | 0.030 | - | - | 1455 | 55 | 40 | - | - | 0.077 | 0.066 | 0.064 | - | - |
| II | 0.078 | 0.075 | 0.068 | 0.065 | - | 1455 | 55 | 40 | 1440 | - | 0.158 | 0.150 | 0.139 | 0.134 | - |
| III | 0.130 | 0.120 | 0.105 | 0.098 | 0.063 | 1460 | 45 | 40 | 60 | 1440 | 0.260 | 0.244 | 0.215 | 0.196 | 0.129 |
| IV | 0.175 | 0.160 | 0.135 | 0.115 | - | 1445 | 40 | 60 | 45 | - | 0.357 | 0.328 | 0.270 | 0.234 | - |
| V | 0.213 | 0.195 | 0.160 | 0.125 | - | 1450 | 50 | 45 | 40 | - | 0.432 | 0.395 | 0.326 | 0.256 | - |
| VI | 0.268 | 0.233 | 0.185 | 0.140 | 0.080 | 1445 | 45 | 60 | 45 | 1450 | 0.547 | 0.480 | 0.368 | 0.286 | 0.162 |

Table 8. Efficiencies and Volumetric Capacities.

| Fan | Total pressure efficiency $r_{1 p}$ by ( 18 ), in $\%$. |  |  |  |  | Effective pressure efficiency $\eta p^{\prime}$ by (19) in \%. |  |  |  |  | Total mechanical efficiency $n_{m}$ by (16) in \%. |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 |
| 0 | 82 | 74 | 68 | 60 | 49 | 82 | 74 | 68 | 60 | 49 |  |  |  |  |  |
| I | 75 | 69 | 63 | 59 | 52 | 72 | 66 | 60 | 54 | 43 | 68 | 65 | 56 | - | - |
| II | 74 | 76 | 74 | 64 | 53 | 65 | 64 | 58 | 43 | 29 | 65 | 71 | 73 | 61 | - |
| III | 79 | 80 | 73 | 65 | 49 | 62 | 55 | 45 | 32 | 17 | 71 | 73 | 68 | 59 | 46 |
| IV | 82 | 77 | 71 | 64 | 48 | 49 | 40 | 30 | 20 | 9 | 72 | 70 | 71 | 60 | - |
| v | 76 | 74 | 68 | 61 | 46 | 32 | 24 | 17 | 11 | 5 | 72 | 68 | 63 | 56 | - |
| VI | 68 | 67 | 63 | 59 | 45 | 11 | 8 | 5 | 3 | 2 | 62 | 57 | 57 | 55 | 40 |

Table 8. Continued.

| Fan <br> resistance. | Effective mechanical efficiency $\eta_{m_{m}}^{\prime}$ by ( 17 ) in $\%$. |  |  |  |  | Volumetric Capacity $\eta_{v}$ by (20). |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 1 | 2 | 3 | 4 | 5 | 1 | 2 | 3 | 4 | 5 |
| 0 |  |  |  |  |  |  |  |  |  |  |
| I | 65 | 63. | 54 | - | - | 0.78 | 0.86 | 0.96 | 1.30 | 1.82 |
| II | 58 | 59 | 57 | 41 | - | 1.45 | 1.77 | 2.12 | 2.56 | 3.06 |
| III | 46 | 51 | 41 | 29 | $16^{\prime}$ | 2.24 | 2.62 | 2.88 | 3.32 | 3.60 |
| IV | 45 | 36 | 29 | 18 | - | 3.01 | 3.32 | 3.60 | 3.93 | 4.05 |
| v | 30 | 22 | 16 | 10 | - | 3.60 | 3.88 | 4.05 | 4.21 | 4.17 |
| VI | 10 | 7 | 4 | 3 | - | 4.20 | 4.28 | 4.37 | 4.44 | 4.26 |

FLATE 1. Characteristics of fan 1.

$\rightarrow Y_{r}$ in $\mathrm{ml}^{3} / \mathrm{s}$

PLATE 2. Characteristics of fan 2.


PLATE 3. Characteristics of fan 3.


PLATE 4. Characteristics of fan 4.

$\rightarrow V_{r}$ in $\mathrm{m}^{3} / \mathrm{s}$

PLATE 5. Characteristics of fan 5.


PLATE 6. Summary of curves D and $P$


PLATE 7. Summary of curves $E^{\prime}$


PLATE 8. Summary of curves $E$ and $E_{f}$


PLATE 9. Summary of curves $P$


PLATE 10. Summary of curves $H_{w}$


PLATE 11. Summary of curve $\eta_{m}$

Vn in min

PLATE 12. Summary of corves $\boldsymbol{\eta}^{\prime}{ }_{m}$


PLATE 13. Summary of curves $\eta_{p}$


PLATE 14. Summary of curves $\eta^{\prime}{ }_{p}$


PLATE 15. Summary of curves $\eta_{v}$



[^0]:    1) Charies H. Innes, "The Fan". 259 (1919)
    T. Suhara, "Photographic Study on the flow of air through a centrifugal fan". XXIV, 66 (1922).
[^1]:    2) Ostertag, "Kolben-und Turbokompressoren" 217 (1919)
    3) Gramberg, "Technische Messungen", III (1923)
[^2]:    5) Gramberg, "Tech. Messungen" in2
    6) "Hütte" zo Auf, I, 313
[^3]:    7) Watson, "Practical Physics," 248 (1917)
    8) "Hütte" zo Auf. I. 322
[^4]:    9) "Hütte" 20 Auf. I. 297
[^5]:    14) Ihering, " Die Gebläse " 550 (1903)
[^6]:    15) Ostertag, "Kolben-und Turbokompressoren ", 155
[^7]:    16) Gramberg, "Tech. Messungen" 3 x 8.
